

Extended Desert Calculation Results With Comparisons to PRISCILLA Experimental Data and a Near-Ideal Calculation

Charles E. Needham Robert G. Ekler Lynn W. Kennedy

ARL-CR-235

July 1995

prepared by

S-Cubed, a Division of Maxwell Laboratories, Inc. 2501 Yale Boulevard, SE, Suite 300 Albuquerque, NM 87106

under contract

DAAL01-94-P-1217



APPROVED FOR PUBLIC RELEASE; DISTRIBUTION IS UNLIMITED.

DYIG QUALITY INSPECTED 5

19950830 082

NOTICES

Destroy this report when it is no longer needed. DO NOT return it to the originator.

Additional copies of this report may be obtained from the National Technical Information Service, U.S. Department of Commerce, 5285 Port Royal Road, Springfield, VA 22161.

The findings of this report are not to be construed as an official Department of the Army position, unless so designated by other authorized documents.

The use of trade names or manufacturers' names in this report does not constitute endorsement of any commercial product.

REPORT DOCUMENTATION PAGE

Form Approved
OMB No. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.

Davis riight to y				
1. AGENCY USE ONLY (Leave blank)	2. REPORT DATE 3. REPORT TYPE A		D DATES COVERED	
	July 1995	Final, Jan 94-		
4. TITLE AND SUBTITLE			5. FUNDING NUMBERS	
Extended Desert Calculation Re	sults With Comparison	s to PRISCILLA	C: DAAL01-94-P-1217	
Experimental Data and a Near-I	deal Calculation		4G061-415-U2	
Exponitional Data and a real				
6. AUTHOR(S)			4G061-515-U2	
Observation F. Manadham Bahard C.	Elder and Lunn W. K.	annody		
Charles E. Needham, Robert G.	ERIEF, AND LYTHI W. NO	STRIEGY		
7. PERFORMING ORGANIZATION NAME			8. PERFORMING ORGANIZATION	
S-Cubed, a Division of Maxwell	Laboratories, Inc.		REPORT NUMBER	
2501 Yale Boulevard, SE, Suite			SSS-DFR-94-14920	
Albuquerque, NM 87106				
9. SPONSORING/MONITORING AGENCY	NAME(S) AND ADDRESS(ES)	10. SPONSORING / MONITORING AGENCY REPORT NUMBER	
U.S. Army Research Laboratory			AGENCY REPORT HOWBER	
ATTN: AMSRL-WT-NC				
Aberdeen Proving Ground, MD	21005-5066		ARL-CR-235	
Abeldeen Floving Glodika, MD	£1005-5000			
			<u> </u>	
11. SUPPLEMENTARY NOTES	et is Dishard E. Lattow	LIC Army Dococrob	Laboraton, ATTN: AMSRL-	
The point of contact for this repo				
· · · · · · · · · · · · · · · · · · ·	ind, MID 21005-5000.	Computer time supplie	ed by Headquarters, Deletise	
Nuclear Agency. 12a. DISTRIBUTION / AVAILABILITY STA	FEMENT		12b. DISTRIBUTION CODE	
12a. DISTRIBUTION / AVAILABILITY STA	ENICHI		125. Bistinbotton Cost	
Approved for public release; dis	ribution is unlimited			
Approved for public release, disc	Houton is unimited.			
13. ABSTRACT (Maximum 200 words)			<u> </u>	
An extended calculation of the	ne non-ideal airblast er	vironment resulting fro	m the PRISCILLA nuclear	
detonation has been completed. experimentally determined them	This calculation used	the most recent, acce	pted interpretation of the	
experimentally determined them	nal layer model.1 The	calculation included the	e effects of turbulence, surface	
roughness, and dust sweep-up i	n determining the near	-surface blast environr	nent. Full hydrodynamic	
definition of the precursor enviro	nment is now available	e from ground zero to a	a distance of over two	
kilometers. Information includes waveforms at over 1,000 locatio	Tuli spatial definition a	i selected times (abou	t 25) and tull-time, resolved	
show good to excellent agreement	iis. Tile lesuits of the C ant in all measured par	arculation are compan amotore	eu with expenimental data and	
SINW GOOD TO excellent agreeme	mit in all incasuleu paid	ameters.		
An accompanying coloulation	without a thormal law	or was also extended t	o over a two kilometer range	

An accompanying calculation without a thermal layer was also extended to over a two-kilometer range. This calculation served as the "ideal" case. The "ideal" calculation included the effects of surface roughness and turbulence but not an interaction with a thermal layer or dust sweep-up. Results of this calculation are used to quantify the differences specifically caused by thermal and dust interactions.

The excellent agreement between experiment and calculation demonstrates the degree of understanding of the physics involved in blast propagation over real surfaces. This understanding of the free-field environment is the necessary first step to predicting loads and response of vehicles or other targets subjected to such an environment.

14. SUBJECT TERMS	15. NUMBER OF PAGES 108			
non-ideal blast, dynamic pressure impulse, nuclear blast, airblast, dust, turbulence, thermal precursor		, unduct, edot,	16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT	18. SECURITY CLASSIFICATION OF THIS PAGE	19. SECURITY CLASSIFICATION OF ABSTRACT	20. LIMITATION OF ABSTRACT	
UNCLASSIFIED	UNCLASSIFIED	UNCLASSIFIED	UL	

INTENTIONALLY LEFT BLANK.

FOREWORD

This work was performed for the U.S. Army Research Laboratory (ARL) under Contract DAAL01-94-P-1217. The calculations were made using the latest version of the S-Cubed Hydrodynamic Advanced Research Code (SHARC). This code has been upgraded to include a version of a K-ε turbulence model, which has been modified by S-Cubed² for non-steady, compressible fluid flow. The turbulence model has a rough law of the wall boundary layer model³ and a dust sweep-up model,⁴ both of which were used for the desert calculation. The K-ε model and the rough law-of-the-wall were also used in the near-ideal calculation. It is the combination of high-order differencing, efficient computer algorithms, and realistic physical models that has made the agreement with experimental data possible.

We would like to acknowledge the efforts of Rich Lottero, Klaus Opalka, and Bud Raley of ARL for making this work possible, and John Keefer and Noel Ethridge of ARA for their guidance in matters of thermal layer development and experimental data interpretation.

Accesio	n For			
NTIS DTIC Unanno Justific	TAB ounced	N		
By				
Availability Codes				
Dist	Avail at Spec			
A-1				

INTENTIONALLY LEFT BLANK.

TABLE OF CONTENTS

		Page
	FOREWORD	iii
1.	BACKGROUND	1
2.	INITIAL CONDITIONS	2
3.	CALCULATED IDEAL RESULTS	4
4.	CALCULATED DESERT SURFACE RESULTS	5
5.	COMPARISONS OF CALCULATIONS WITH EXPERIMENTAL DATA	8
6.	CONCLUSIONS	11
	REFERENCES	13
	APPENDIX A: PARAMETER SUMMARY PLOTS	A -1
	APPENDIX B: WAVEFORM COMPARISONS	B-1
	APPENDIX C: HYDRODYNAMIC PARAMETERS AS A FUNCTION OF HEIGHT FOR SELECTED GROUND RANGES	C-1
	APPENDIX D: CONVERSION TABLE	D-1
	DISTRIBUTION LIST	DIST-1

INTENTIONALLY LEFT BLANK.

SECTION 1 BACKGROUND

This calculation is the product of over four decades of research into thermally-precursed airblast. It has been made possible by significant advances in numerical differencing techniques, physical modeling development, and computer hardware improvement. The importance of turbulence and a good boundary layer model were demonstrated during the DIAMOND ARC experiments in 1989⁵.

The role of pre-shock dust has been debated for many years. The most recent calculations assume that pre-shock dust loading of the air is negligible. This is supported by several different types of observations. First, measurements of the temperature and sound speed as a function of height above desert surfaces clearly show that the peak temperature occurs near the ground and decreases rapidly with height above the surface. If significant dust had been lofted, this dust would have absorbed incident radiation and caused heating of the air well above the ground. Second, photography from cameras located over two kILOMETERS from bus and truck targets⁷ clearly show the buses and trucks until shock arrival. This means that the visible light, mean free-path is in excess of one kilometer and therefore, no significant absorption of radiated energy can occur in the three-meter depth of the observed dust height. Third, shock photography indicates that the precursor shock is linear and extends to very near the ground. If there were any significant enhancements of sound speed above the ground, the precursor shock would reflect the structure of the thermal layer with the front traveling faster in the highest sound speed region. This is further evidence that the thermal layer is hottest near the ground and cools rapidly with height. Fourth, arrival times measured by free-field gages and gages on structures⁸ show that the signal arrival at ground level is earlier than at the three-foot elevation. Some gages on structures were only six inches above the surface. Even these gages show arrival after the ground-level gages. The shock is traveling faster at ground level than at six inches above the ground. The implication is that the hottest part of the thermal layer is less than six inches thick and that temperature decreases rapidly above that height. All of this points to negligible dust lofting prior to shock arrival.

One continuing subject for study is the observed fact that a precursor signal can and does travel subsonically at ranges beyond the 30-psi ideal overpressure range. This observation has not been fully utilized in this calculation and could lead to even better agreement with experiment at distances greater than one kilometer. The DIAMOND ARC arrival time data⁵ indicate that the leading measured disturbance is traveling below the sound speed in the helium bag by about 30 % for all ranges beyond the 30 psi range. The implication is that sound speeds in the pre-shock thermal layer may be 30 to 50 percent greater than the measured signal velocity at ranges greater than 3500 feet for PRISCILLA. This phenomenon has been studied and reported in Reference 10 by Barthel at S-CUBED. Such an extended hot layer has not yet been used in a calculation.

SECTION 2 INITIAL CONDITIONS

The calculations described in Reference 9 were used as initial conditions for the extended calculations reported here. The "ideal" calculation was started from a time of two seconds and run to a time of four seconds.

The desert thermal layer calculation was restarted when the shock reached about 1,450 feet in ground range. The thermal layer temperature was increased to match a sensible upper bound of the experimentally-determined pre-shock thermal layer temperature and was run to a distance of 3,000 feet. The calculation indicated a shorter precursor prior to "main shock" arrival at the 3,000-foot range than was seen in the experiment, indicating that the thermal layer used in the calculation was not as hot as that which existed in the experiment. The thermal layer temperature for distances beyond 3,000 feet was re-evaluated, based on measurements interpreted from several nuclear events. The thermal layer was extended to a range of 4,300 feet. The calculation was restarted a second time at a distance of 2,500 feet and continued to four seconds and a distance of 6,000 feet. Figure 1 shows the three temperature/sound speed curves used for the three calculations. The results given in this report used the maximum of the three curves at any given distance.

Both calculations (ideal and precursed) used the S-CUBED K- ϵ turbulence model. This model is an extension of the usual K- ϵ model which uses a variable coefficient for formation and dissipation of turbulence, based on local conditions and the history of the flow. The S-CUBED modifications extend the K- ϵ model to compressible, non-steady flows. Both calculations used a law-of-the-wall for real surfaces in conjunction with the turbulence model. The ideal calculation used a smooth wall Clauser law-of-the-wall and the desert used a smooth wall Rubesin law-of-the-wall to represent the surface interaction.

The ideal calculation used a shock-following subgrid with 10-centemeter zones throughout most of the calculation. The precursor calculation used a similar shock following subgrid, but had 30-centemeter zones for most of the calculation duration.

THERMAL LAYER SOUND SPEED

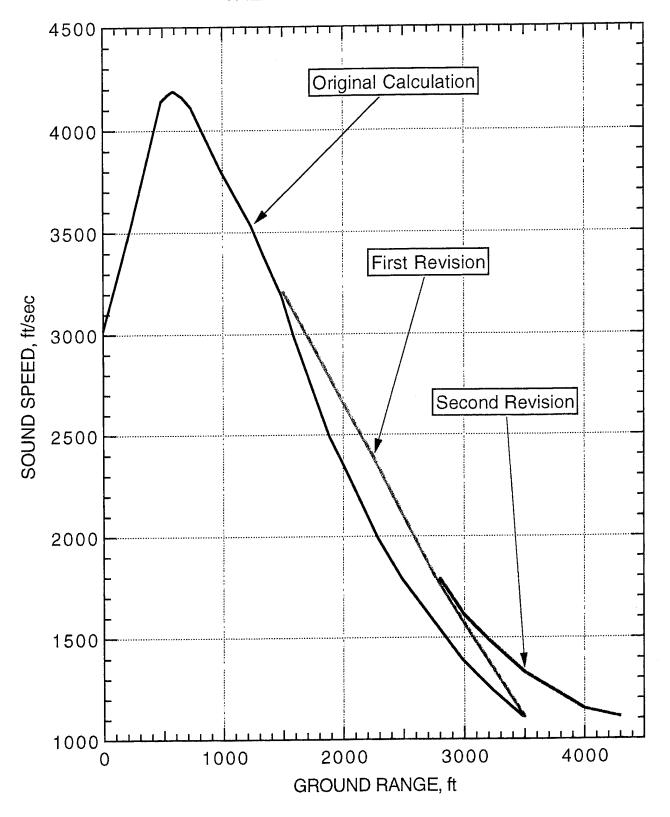


Figure 1. PRISCILLA sound speed vs. range.

SECTION 3 CALCULATED IDEAL RESULTS

The four-second problem time carried the shock front beyond the 6,000-foot range to an overpressure of just under 4 psi. The overpressures measured on PRISCILLA, beyond 4,000 feet, are in good agreement with the results of the ideal calculation. The overpressure positive phase was complete at the 5,000-foot range (about 5 psi) and the impulse agrees well with the experimental value at that distance. Because the positive phase was not complete beyond the 5,000-foot range, impulse comparisons could not be made in this region. Peak overpressures and waveforms for the ideal calculation for distances less than 3,000 feet were reported in Reference 9 and will not be repeated here, except as a reference for the desert calculation results.

A simple method was developed to remove the numerical overshoot observed on nearly all of the ideal waveforms. The overshoot peak was averaged with the first undershoot following the overshoot using the geometric mean. This method was developed under a previous contract and has been checked using results from several more complex interpolation methods. The geometric mean results have been found to be within acceptable error bounds, although perhaps a few percent high in some cases.

Summary plots of arrival time, overpressure, overpressure impulse, dynamic pressure, and dynamic pressure impulse are contained in Appendix A for the results of the ideal calculation. These results are compared to the results from the desert calculation and to experimental data from the PRISCILLA event. Waveform comparisons at a number of selected ranges are included in Appendix B. The waveforms are compared to the desert calculation results and to experimental waveforms where possible.

We have also included a number of parameter versus height plots at selected ground ranges. These extend from ground level to 40 feet above the ground. Comparisons are made with the results from the thermal layer calculation. These plots are included in Appendix C. Overpressure, arrival time, and impulse for the ideal calculation show very little variation with altitude. At large distances, beyond 4,000 feet, the rough surface has a small effect in reducing the near surface dynamic pressure.

SECTION 4 CALCULATED DESERT SURFACE RESULTS

A summary of the initial conditions is given in Section 2. Because the same thermal layer was used for this calculation as that reported in Reference 9, the results for distances less than 2,500 feet are the same as those of Reference 9 and will not be discussed here.

The calculation was carried to a time of four seconds and a distance of just over 6,000 feet. At the end of the calculation, the positive duration of the overpressure and dynamic pressure were complete for all ranges having overpressures greater than or equal to five psi.

During the calculation, we found that the station positions moved from the positions assigned as part of the initial conditions. We were able to trace this motion to the limitations of the 32-bit work station on which the desert calculation was made. During each rezone, the stations were assigned a new cell index position and this was converted back to a real distance. Because of the limited accuracy of 32-bit floating point arithmetic, many of the stations moved a small amount after each rezone. Several hundred rezones were completed during the progress of this calculation. Stations at larger ranges were effected more than those near ground zero.

The results reported in the summary plots of Appendix A have used the positions of the stations at the time of shock arrival. This problem does not effect the results of the ideal calculation because it was run on the Cray, a 64-bit machine. It also did not effect the vertical positions of the stations because few rezones occurred in the vertical direction. The vertical distances were small compared to the horizontal, and new zone indices were not calculated as often.

The dynamic pressures reported are the result of both air and dust contributions. The dust contribution has been assigned a "registry coefficient" of 0.5. The dust and air were treated as fluids and dynamic pressure was calculated as:

$$DP = 0.5 * rho * u * lul,$$
 (1)

where rho is the total density (air plus dust) of the zone.

The arrival time curves on the first figure in Appendix A show that the precursor separates from the ideal at a distance of less than the height-of-burst and remains ahead of the ideal arrival throughout the two-kilometer ground range. The maximum separation between precursor arrival and ideal is just over 300 feet at a time of 0.6 seconds. The waveforms of Appendix B show that the precursor arrives before the ideal at ranges as small as 350 feet.

The summary plots of Appendix A show that the maximum overpressure in the desert thermal case is as little as one-third of the ideal overpressure. The overpressure at the precursor front may be more than an order of magnitude less than the peak pressure occurring later in the waveform. The overpressure impulse differs by less than ten percent from the ideal over the entire range of comparison.

The maximum dynamic pressure is, at some ranges, as much as a factor of three greater than the ideal. The dynamic pressure curves cross near 3,000 feet and the precursor peak dynamic pressure is less than that of the ideal for all greater ranges. The dynamic pressure impulse exceeds the ideal by as much as a factor of eight between

ground ranges of 2,000 and 3,500 feet, then falls below the ideal values at greater ranges.

The waveforms of Appendix B show the details of many of the features described above. At a range of 2,500 feet, the ground-level overpressure waveform has a rounded front, with the peak overpressure near the front. Only a single peak is evident. The peak overpressure is about one-third of that for the ideal calculation. Overpressures at three and ten feet are very close to those at ground level. The dynamic pressure waveform shows that the maximum pressure occurs in a secondary peak some 300 milliseconds behind the precursor wave. The peak is about two and one-half times the ideal peak.

At a range of 3,000 feet, the overpressure waveforms are similar but the non-ideal peak is about half that of the ideal. The increase relative to the ideal is a sign of the precursor front slowing and decreasing in extent. The dynamic pressure waveform at this range shows two major rounded peaks, with the second occurring about 150 milliseconds after arrival. The peak is about twice that of the ideal. The decrease in separation time of the two peaks beyond 2,500 feet is a further indication of the beginnings of precursor clean-up.

By 3,500 feet, the calculation shows a rounded front, an inflection, and a sharp rise to a peak overpressure which is about 75 percent that of the ideal. The sharp rise to the second peak is an indication of precursor clean-up. The dynamic pressure waveform shows multiple peaks and a slow rise after first arrival. The peak dynamic pressure is comparable to the ideal peak. The decay after the peak is reached is much slower than in the ideal case and leads to a dynamic pressure impulse of about three times the ideal

At a range of 4,000 feet, the overpressure waveform is nearly ideal. The major difference between precursed and ideal at this range is that the peak overpressure is a few percent lower than the ideal. The dynamic pressure waveform shows that the precursed waveform falls below the ideal at all times, thus causing a lower dynamic pressure impulse.

For ranges beyond 4,000 feet, the non-ideal waveforms are very close to the ideal but the arrival times are somewhat smaller and the peaks remain a few percent lower than ideal.

Appendix C contains comparisons of various parameters as a function of height at selected ground ranges. The plots cover the variation with altitude from ground level to 40 feet above the ground. At the 2,100-foot ground range, the peak precursor overpressure is about one-third that of the ideal with the near ground-level pressure as much as 50 percent greater than that above 10 feet in altitude. The ideal does not vary with altitude to less than one percent.

At 2,300 feet, some variation in peak overpressure is seen in the lower 10 feet, but the variations are less than 30 percent in the precursor case. In general, the precursed maximum overpressures are about one-third those of the ideal. The ideal shows no variation with altitude.

The comparison at 2,550 feet shows the precursor pressures to be less than half those of the ideal case. Variations with altitude are less than ten percent for the precursor and less than one percent for the ideal. This trend continues through the 2,950-foot ground range.

The temperature and sound speed in the thermal layer decrease rapidly beyond a range of 3,000 ft. This marks the clean-up phase of precursor propagation. The variation with height at 3,250 and 3,650 shows dramatic changes as the layer cools. At

3,250 feet the peak overpressure at 40 feet above the surface is about 30 percent greater than near the surface, but is still about 75 percent of the ideal. The ideal remains unchanged with height. By 3,650 feet, the overpressure above 20 feet differs from the ideal by less than 10 percent, while near ground level the overpressures are about 80 percent of ideal.

The thermal layer terminated at the 4,300-foot range; no pre-shock heating was present beyond this range. The variations with height beyond the end of the thermal layer are caused by residual differences in energy distribution in the shock and transient flows which are attempting to equilibrate along the shock front. Variations in height are small, of the order of two percent, and the differences between precursed and ideal are less than ten percent.

The arrival time as a function of height plots show no surprises. The curves are very smooth and show that the arrival at ground level is earlier than at any other height. This is in agreement with observed arrival times on structures from the PRISCILLA event. The precursor arrival times are earlier than the ideal for all ground ranges. Beyond the 4,300-foot range, the arrival time does not change with height.

The dynamic pressure plots of Appendix C show that the dynamic pressure nearest ground level is about a factor of two higher than at an elevation of three feet at the 2,100-foot ground range. This characteristic decays rapidly as the boundary layer grows behind the precursor front. By 2,500 feet, the maximum dynamic pressure occurs three to six feet above the ground. For all ground ranges less than about 3,200 feet, the precursed dynamic pressure exceeds that of the ideal near ground level. At 2,950 feet, the dynamic pressure near ground level is more than twice the ideal but falls below the ideal above 35 feet from the surface.

As with the overpressure, several oscillations are present in dynamic pressure as the precursor cleans up. Apparently, energy is exchanged between dynamic pressure and overpressure as the shock front adjusts to the absence of a thermal layer.

The most dramatic effect is seen in the dynamic pressure impulse. At a range of 2,100 feet, the near-surface dynamic pressure impulse from the precursor calculation exceeds the ideal by more than an order of magnitude, while at the three-foot elevation, the ideal is exceeded by about a factor of five. The impulse remains greater than the ideal for all heights. Some effect of the boundary layer can be seen in the reduction of dynamic pressure impulse for the ideal case also.

The effect of the boundary layer is evident in the 2,300-foot plot. The impulse for the precursed calculation changes from being greatest at ground level to being reduced such that the maximum occurs near the three-foot height. The ideal impulse is also reduced near ground level.

The maximum impulse of the precursor is greater than the ideal at all heights, but approaches the ideal near the 40-foot height throughout the clean-up phase, to a distance of nearly 4,000 feet. The impulse drops sharply beyond 4,000 feet and falls below the ideal for the remainder of the calculated range.

SECTION 5 COMPARISONS OF CALCULATIONS WITH EXPERIMENTAL DATA

The summary plots of Appendix A contain comparisons of calculations with nearly all available data from the PRISCILLA event.

The arrival time curve shows excellent agreement with the majority of the data. Some data have been questioned (Keefer private communication) because of the type of instrumentation used. In general, the desert calculation shows good agreement with the measured data and its wave front always arrives earlier at any given range than does that of the ideal case. The data indicates a faster propagation at ranges beyond 3,500 feet than was calculated. This is one of many indications that a significant thermal layer existed well beyond the 4,000-foot range in the PRISCILLA test.

The overpressure summary plot includes experimental data from ground level, three-foot and ten-foot heights. The three- and ten-foot elevation data agree better with the ideal overpressures than with the precursor values. The calculated overpressures are for ground level only. Two overpressures are plotted for each calculation: the first peak and the maximum. The only range for which these curves differ in the ideal case is during double and complex Mach reflection. This limited region extends from about 600 to 1,400 feet. For the precursor calculation, the peak overpressure falls below the ideal almost immediately. As the precursor forms and generates a double peaked waveform, the two curves diverge. At a range of 350 feet, only one peak is present, but by 500 feet a weak shock having a peak of about 10 percent 1 of the maximum leads the so-called "main wave". The precursed overpressure peaks fall below those of the ideal to a range of just over 4,000 feet, the end of the thermal layer. The first peak may be as little as 10% of the maximum overpressure at a given range.

The calculated precursor overpressure is in good agreement with all of the experimental data. It should be noted that the overpressure reaches a relative minimum at a range of just over 2,500 feet, then rises to a relative maximum at about 4,000 feet. This maximum is slightly higher than the ideal at this range. The peak then falls back to the ideal level for the remainder of the calculated ranges. This behavior is in agreement with the experimental data from several nuclear shots, including PRISCILLA.

The increase in overpressure as a function of ground range, beyond the 2,500foot range, has been observed experimentally and is now confirmed by calculation. The rise and fall of the overpressure with range leads to a triple valued function for the range of a given overpressure; e.g., there are three ranges at which 8 psi occurs. The calculation indicates 2,200, 3,200, and 4,100 feet all had a peak overpressure of 8 psi. This triple valued function is the cause of the non-ideal height-of-burst curves having loops and multiple values as a function of ground range and height of burst. These characteristics are real, calculable, and we believe that we now understand them.

The overpressure impulse data have considerably more scatter than the peaks. The calculations fall near the high side of the data. The causes for this scatter can be seen in the waveforms of Appendix B. Some waveforms fall below ambient at a relatively early time after shock arrival, while others do not return to ambient for an extended period. Such scatter is an indication of the difficulty of making measurements in the nuclear environment and the variety of waveforms measured at the same ground

¹ The agreement of the computation with the BRL waveform at 1650 feet is apparently fortuitous. The timing on the BRL waveform is now believed to be in error; the BRL waveform should be expanded so that the maximum peak coincides with that on the SRI peak. The BRL self-recording gages used in PRISCILLA did not have a timing-mark generator.

range. The waveforms depend on the integrated history of the interaction of the shock with the thermal layer, and surface irregularities contribute significantly to variations in this history.

The peak dynamic pressure summary plot shows that the peak measured values differ, in general, by about a factor of two to three from the ideal. The data are above the ideal for ground ranges between 1,200 feet and 3,200 feet. The calculated precursor results show this range of variation and agree with the range at which the dynamic pressure falls below the ideal.

The dynamic pressure impulse data, taken three feet above the surface, are in good agreement with the precursor calculated results. When viewed as in the sixth figure of Appendix A, the data fall onto two lines. The first line is very near the ideal, while the second closely follows the precursor calculation. The data and the calculation indicate that for some ranges the dynamic pressure impulse may exceed the ideal by more than an order of magnitude.

The waveforms of Appendix B include all available desert line waveforms. No effort has been made to edit, delete, or emphasize any particular waveform or comparison. Many of the gages did not have associated arrival times, but times were given as relative to first signal arrival. We have shifted all desert waveforms so that the first signal arrives at the time of the calculated precursor waveform. Because the current calculation differs from that of Reference 9, only at distances greater than 1,450 feet, the discussion of waveform comparisons will be limited to those beyond the 1,450 foot range.

The calculated waveforms of Appendix B represent the mean flow parameters at the positions given. The calculations include the turbulent contribution as a separate parameter. Waveforms using a combination of the mean parameters and the turbulent contribution can be reconstructed from the calculations. This reconstruction includes a full frequency distribution of the Kolmogorov spectrum. The resulting waveforms must then be low-pass filtered to the characteristics of a given gage before comparisons can be made; this has not been done here. The calculated waveforms are therefore somewhat smoother than the data because of the lack of the turbulent component. The turbulence will add oscillations on the waveforms, but impulse values will not be changed.

At 1,650 feet, the agreement between the calculated precursor and the BRL measured overpressure waveforms is excellent. The SRI gage shows greater separation between first and second peaks and a somewhat higher second peak.

At 2,000 feet, the two SRI gages show high, spiking secondary peaks which are as high as those measured at the 1,650-foot range and above the ideal at this range. The calculation shows excellent agreement with the rise and shape of the first peak and agrees with the timing between first and second peaks, but falls well below the second peak pressure. This comparison must be weighed against the measurements taken before and after this ground range.

Some 250 feet further, at 2,250 feet, the BRL waveform indicates no such secondary spike. The calculated precursor waveform is again in excellent agreement with the data.

By 2,500 feet, the second peak is no longer apparent in the waveform. The SRI and BRL gages are in agreement, and the calculated waveform follows the same pattern.

At 3,000 feet, the BRL, SRI, and calculated precursor waveforms are essentially overlays. The two data waveforms show a small spike near 1.25 seconds. The spike

indicates the formation of a secondary shock as precursor clean-up begins. This is not apparent in the calculation.

The clean-up continues, as seen at the 3,500-foot range. The calculation has a shorter, rounded front and a higher second peak than the experimental waveform. This is a further indication that the thermal layer used in this calculation is cooler than existed in the experiment at this range. The experimental waveform falls more quickly after the peak, but this should be compared with the waveform measured just three feet above the ground. Here, the SRI data merges with the calculated waveforms after about 1.7 seconds.

The premature clean-up of the calculation is further demonstrated in the waveforms compared at 4,000 feet. The experiment shows a separation of about 30 ms between the first and second peaks, but the calculation has a single, rapid rise indicates nearly complete clean-up at this range.

SECTION 6 CONCLUSIONS

The results of the "ideal" calculation serve as a benchmark for the definition of the entire airblast flowfield over a realistic surface. This calculation is being and will be used to compare and quantify the effects of dust and thermal layers. The zone size remained at 10 centimeters in the shock following sub-grid to a distance of over 1.2 kilometers. The zone size in the subgrid was then gradually increased to a maximum of 30 centimeters as the shock approached 2 kilometers. The resolution is adequate for this calculation to be considered state-of-the-art.

The desert calculation required some compromise on resolution. The moving subgrid contained zones with dimensions of 30 centimeters throughout the calculation. This compromise was necessary in order to assure completion of the calculation within cost constraints. A desert thermal layer calculation with 10-centimeter resolution at the PRISCILLA scale is still a very desirable goal. This calculation has sufficient resolution to answer many of the questions about thermal layer temperature distribution and the role of dust in the overall flowfield. A higher resolution calculation will require careful reconsideration of the temperature distribution in the thermal layer, the extent of the high sound-speed region, and the consequences of temperature gradients on precursor cleanup.

The desert precursor calculation results, presented here, show the best agreement with experimental data of any calculation by any organization done to date. This comparison includes arrival times, overpressures, dynamic pressures, impulses, and waveform details. We now have defined the flowfield for the PRISCILLA event in sufficient detail to provide high quality environment descriptions, above 25 psi but with less fidelity at lower pressures.

The results of this calculation are being transferred to magnetic media and will be available for further detailed analysis in the future. Some questions about the role of dust versus air in the measurement of dynamic pressure have already been addressed during the comparisons between calculational results and experimental data. The growth of a boundary layer and the interaction of the precursor with the boundary layer can be more fully examined. The role of turbulence in dust lofting and dust distribution behind the precursor is yet to be addressed in detail. Many insights into these associated phenomena and some answers are now available, but further analysis is required to exploit fully this pair of computations.

INTENTIONALLY LEFT BLANK.

REFERENCES

- 1 Carpenter, H.J., Engler, M.J., McCaffree, L.A., "Pre-Shock Thermal Layer Sound speeds Developed from Nuclear Test Data", DNA-TR-89-190, Defense Nuclear Agency, Alexandria Va., August, 1991.
- 2 Pierce, T.H., "Numerical Boundary Layer Analysis with K-E Turbulence Model and Wall Functions," Defense Nuclear Agency Report DNA-TR-87-15, September 1986.
- 3 Barthel, J.R., Needham, C.E., Pierce, T.H., and Schneyer, G.P., "A Computational Model for Precursed Airblasts Over Rough Surfaces," S-Cubed Report SSS-R-89-10003, August 1989.
- 4 Pierce, T.H., "Turbulence and Real-Surface Sub-Models in S-Cubed Hydrocodes," S-Cubed Draft Report DTR-91-12671, 1991.
- 5 Needham, C.E., et. al., "Theoretical Calculations for Precursor Definition," Defense Nuclear Agency Report DNA-TR-90-18, September 1990.
- 6a Swift, L.M., Sachs, D.C., and Kriebel, A.R., "Operation PLUMBBOB, Project 1.3: Air-Blast Phenomena in the High Pressure Region," WT-1043, Stanford Research Institute, Menlo Park, CA, December 1960.
- 6b Bryant, E.J., Keefer, J.H., Swift, L.M., and Sachs, D.C., "Operation PLUMBBOB, Projects 1.8a and 1.8c: Effects of Rough and Sloping Terrain on Airblast Phenomena", WT-1407, Ballistic Research Laboratories, Aberdeen, MD, July 1962.
- 7 Operation PLUMBBOB, Event PRISCILLA, documentary photography, available from DASIAC film archives, Kaman Sciences, Corp., P.O. Box 1479, Santa Barbara, CA 93102-1479.
- 8 Banister, J.R., and Vortman, L.J., "Operation PLUMBBOB, Project 34.1: Effects of a Precursor Shock Wave on Blast Loading of a Structure," WT-1472, Sandia Corporation, Albuquerque, NM, October 1960.
- 9 Crepeau, J.E., Ekler, R.G., Kennedy, L.W., Needham, C.E., and Rogers, S.H., "SHARC Hydrocode Calculations of the PRISCILLA Event," S-Cubed Report SSS-DFR-93-14283, October 1993.
- 10 Barthel, J., "On the Relationship Between Thermal Layer Sound Speed and Precursor Observables," Defense Nuclear Agency Report DNA-TR-88-241, January 1992.

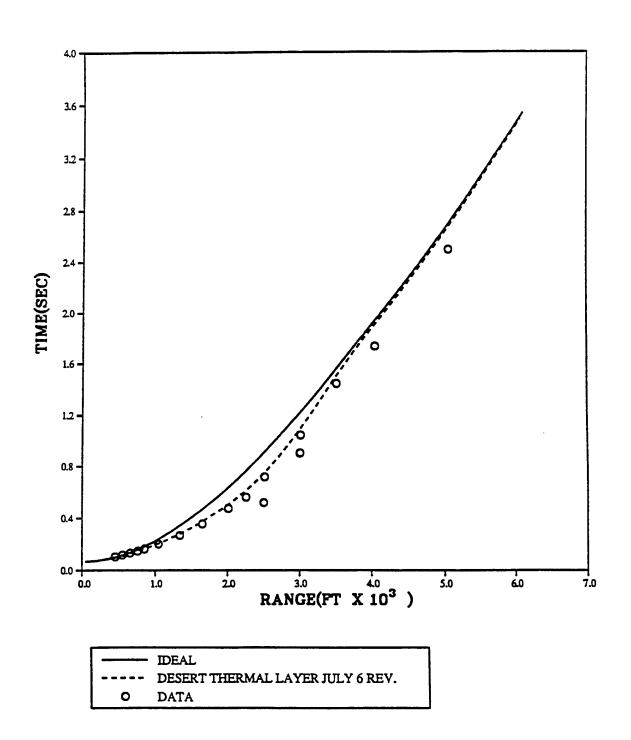
INTENTIONALLY LEFT BLANK.

APPENDIX A PARAMETER SUMMARY PLOTS

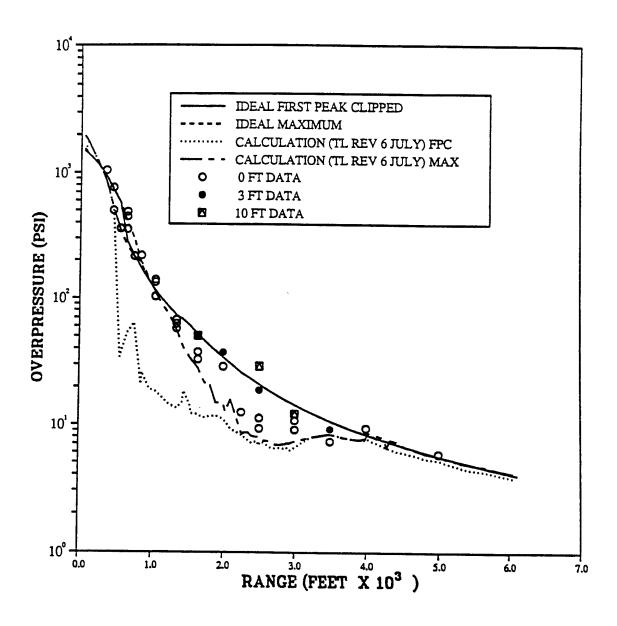
This Appendix contains summary plots of hydrodynamic parameters as a function of ground range. Each plot contains the results of the ideal calculation, the desert calculation, and experimental data. No dynamic pressure measurements were made at ground level. All the experimental dynamic pressure data were taken at least three feet above the surface. Many of the dynamic pressures from the experiment were derived from stagnation pressure measurements at a 3-foot elevation and the overpressure measurements at ground level. The results from the PRISCILLA calculation show that the overpressure varies between ground level and three feet in the region of strong precursor and the assumption of equal overpressures may be in error by 10% or so.

All measured dynamic pressures are taken without regard to the type of gage or its dust registry coefficient. The calculated dynamic pressures include the dust dynamic pressure contribution. In the plots from these calculations, the dust is treated as a fluid and has a registry coefficient of 0.5.

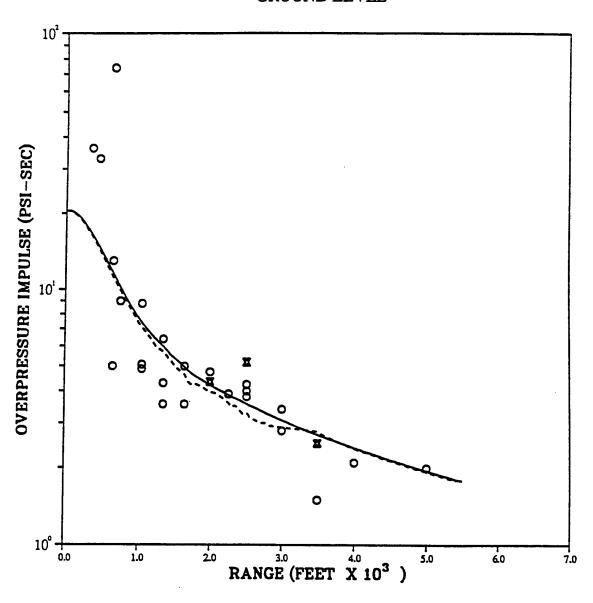
PRISCILLA
ARRIVAL TIME AT GROUND LEVEL



PRISCILLA DESERT OVERPRESSURE AT GROUND LEVEL



PRISCILLA DESERT OVERPRESSURE IMPULSE GROUND LEVEL

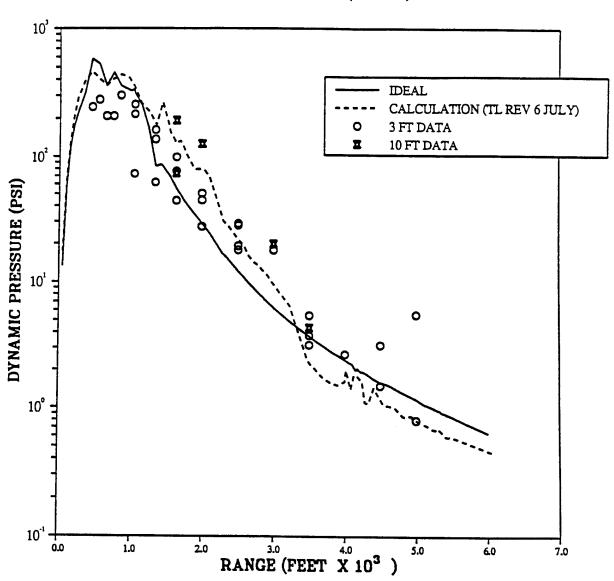


---- IDEAL 0 FT

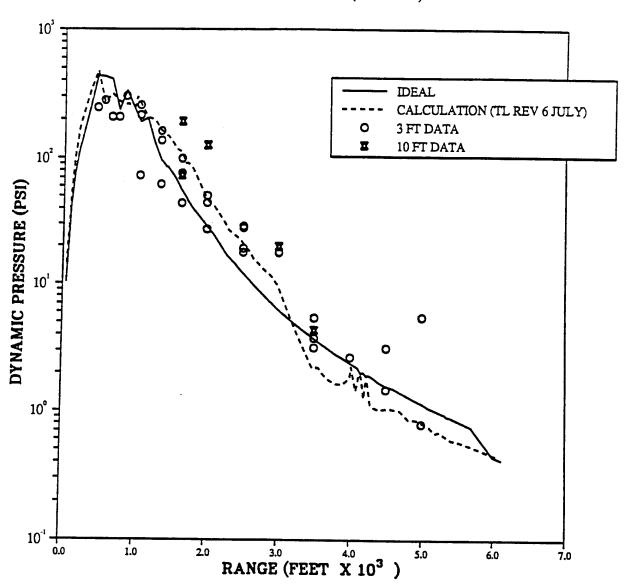
---- CALCULATION 0 FT (TL REV 6 JULY)

O DATA 0 FT
DATA 3 FT

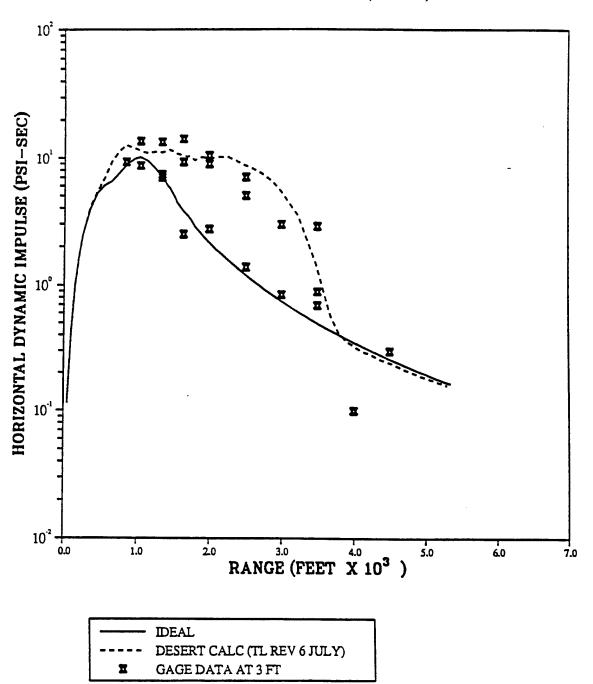
PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS 91.44 CM LEVEL (3 FEET)



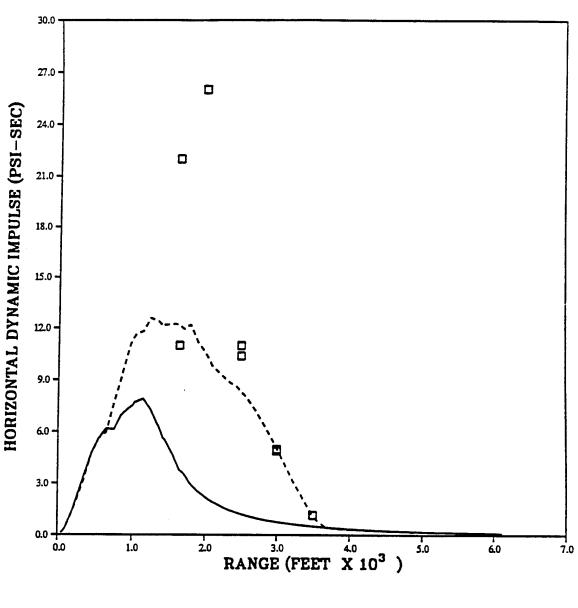
PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS 304.8 CM LEVEL (10 FEET)



DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE AT 91.44 CM LEVEL (3 FEET)



DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE AT 304.8 CM LEVEL (10 FEET)



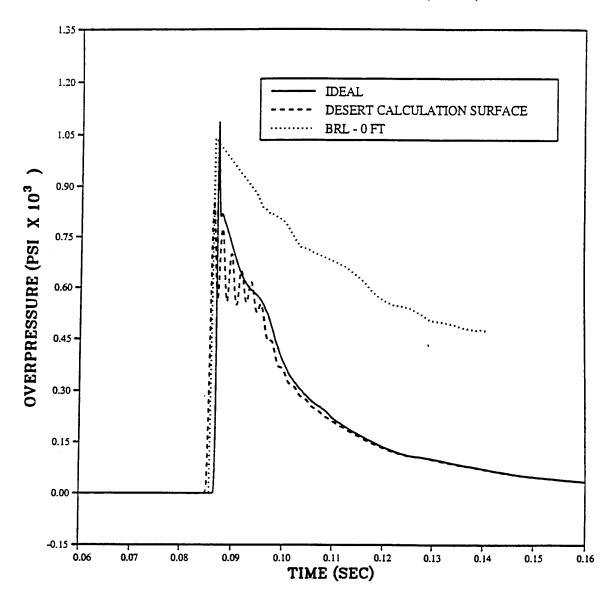
DESERT CALC (TL REV 6 JULY)
GAGE DATA AT 10 FT

APPENDIX B WAVEFORM COMPARISONS

This Appendix contains waveform comparisons of overpressure, dynamic pressure, and their impulses. Each plot contains the ideal waveform, the calculated precursor waveform, and at least one measured waveform at the corresponding distance. Arrival time of the measured waveform has been shifted to agree with the precursor calculation.

More information is available. The dust density as a function of time has been calculated and saved. It is possible to determine the calculated air and dust dynamic pressures independently. Any desired dust registry coefficient or a functional form of the dust registry coefficient may be used. Mach number of the flow as a function of time is also available at any of the station positions. Full descriptions of the turbulence environment are also available at each station, including the turbulent energy and the rate of turbulence dissipation.

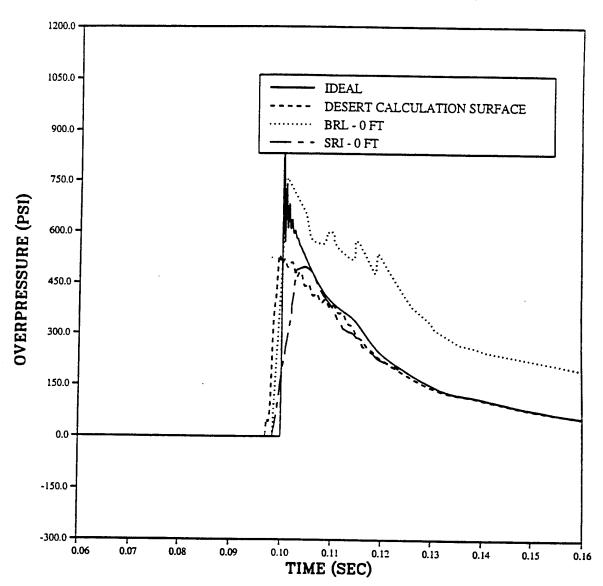
PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 350 FEET (107 M)



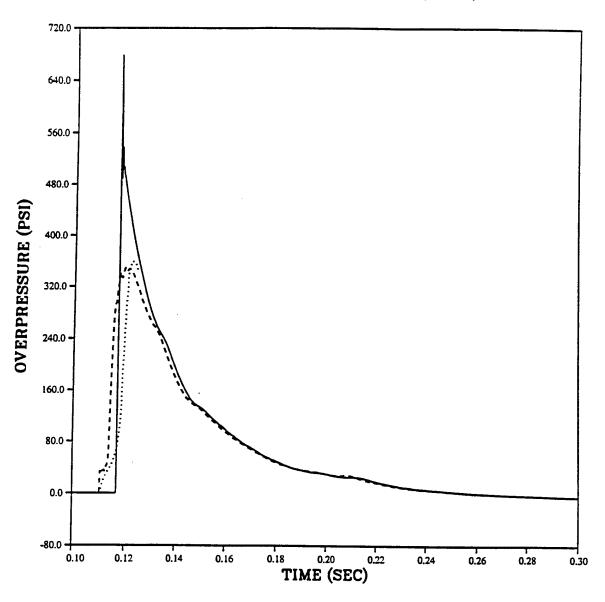
PRISCILLA

CALCULATION - DATA COMPARISONS

OVERPRESSURE AT 450 FEET (137 M)

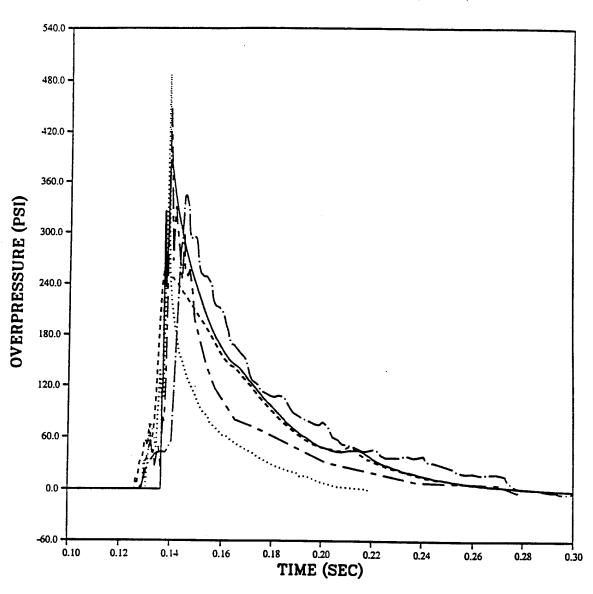


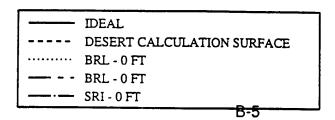
PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 550 FEET (168 M)



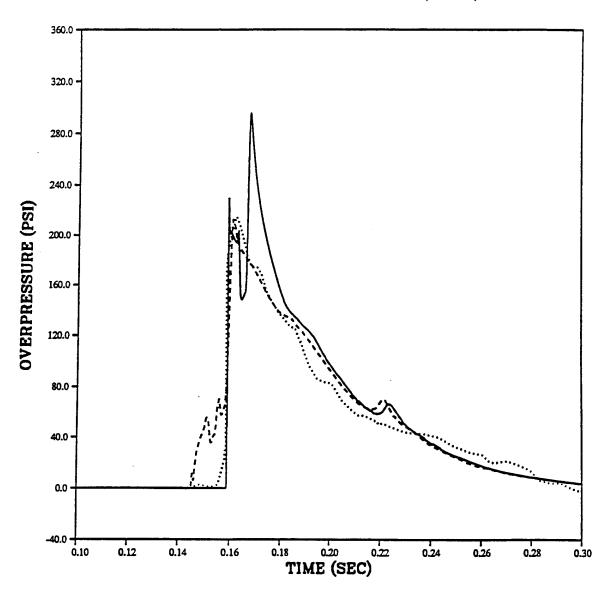
—— IDEAL
---- DESERT CALCULATION SURFACE
..... SRI - 0 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 650 FEET (198 M)





PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 750 FEET (229 M)

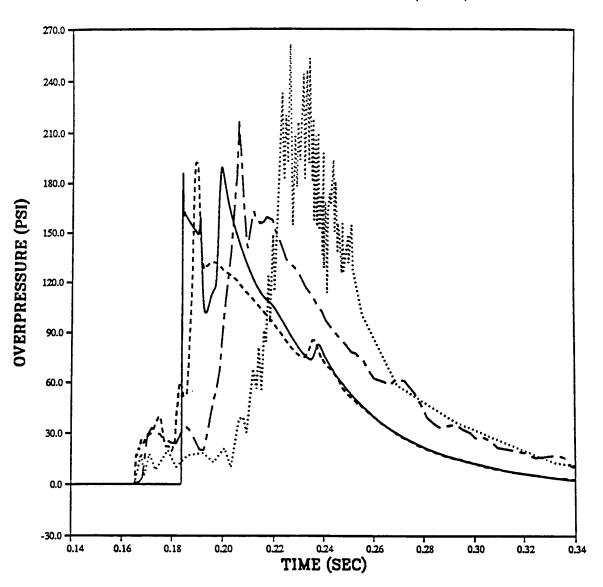


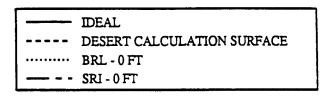
---- IDEAL
---- DESERT CALCULATION SURFACE
...... SRI - 0 FT

PRISCILLA

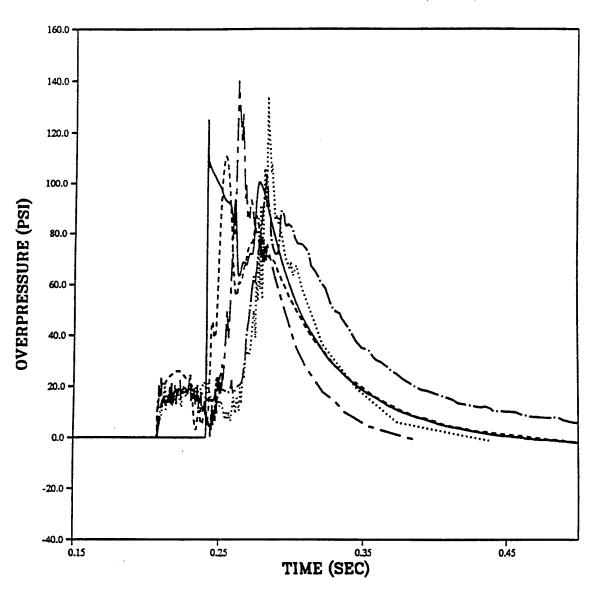
CALCULATION - DATA COMPARISONS

OVERPRESSURE AT 850 FEET (260 M)

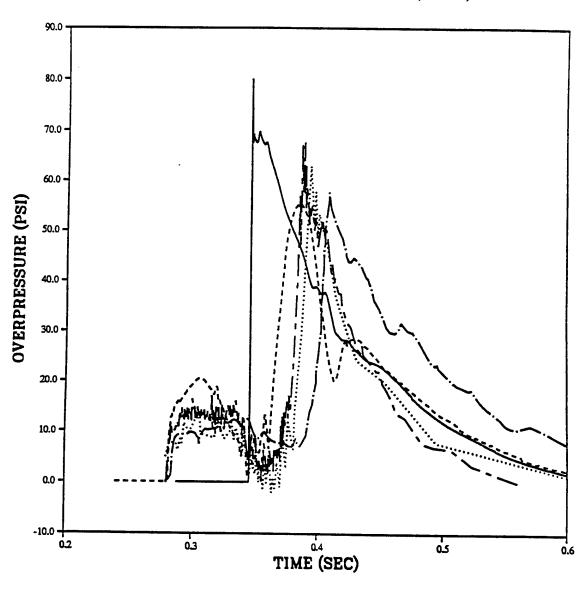




PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 1050 FEET (320 M)



PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 1350 FEET (410 M)

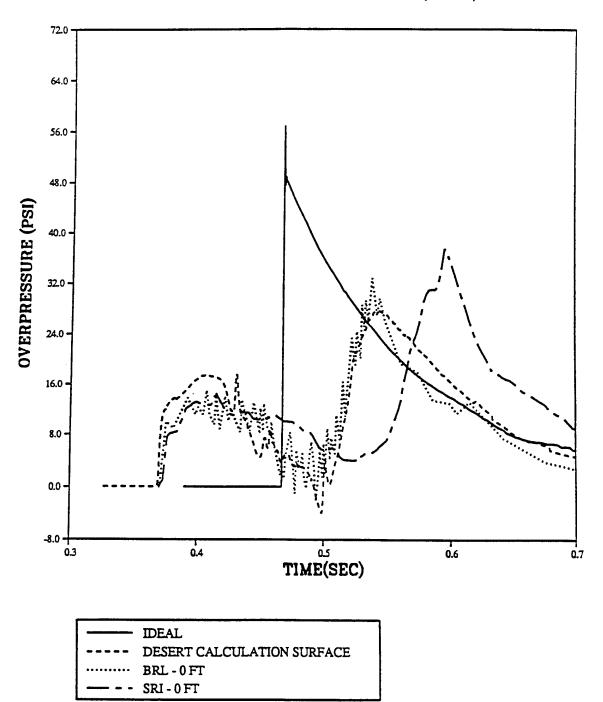


DEAL
DESERT CALCULATION SURFACE
BRL - 0 FT
BRL - 0 FT
SRI - 0 FT

PRISCILLA

CALCULATION - DATA COMPARISONS

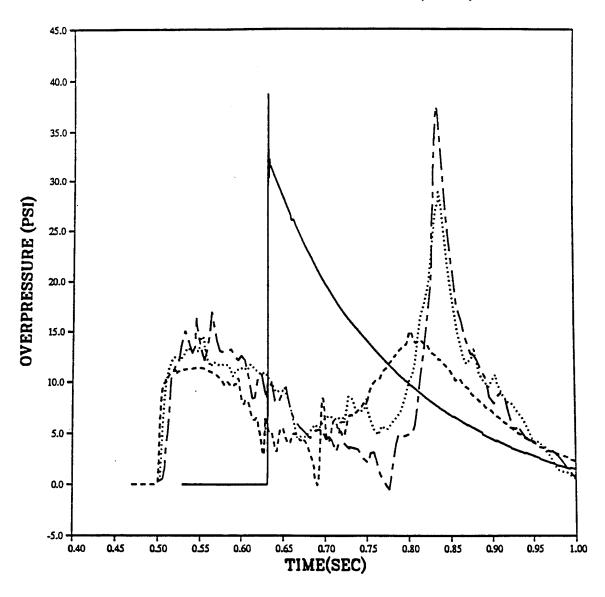
OVERPRESSURE AT 1650 FEET (503 M)



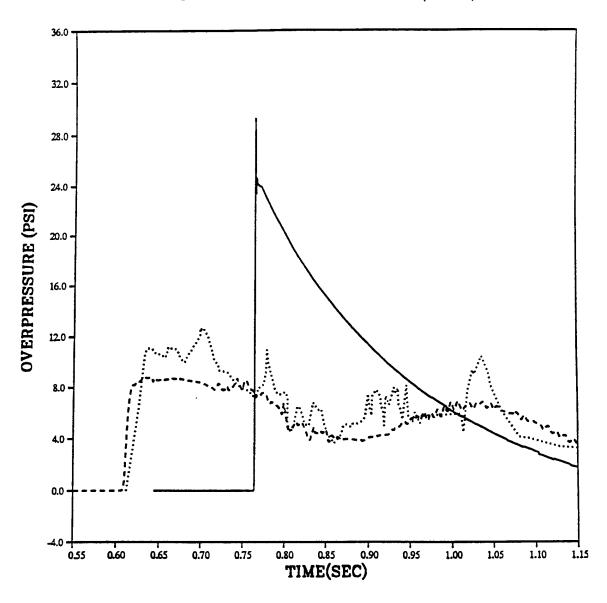
PRISCILLA

CALCULATION - DATA COMPARISONS

OVERPRESSURE AT 2000 FEET (610 M)

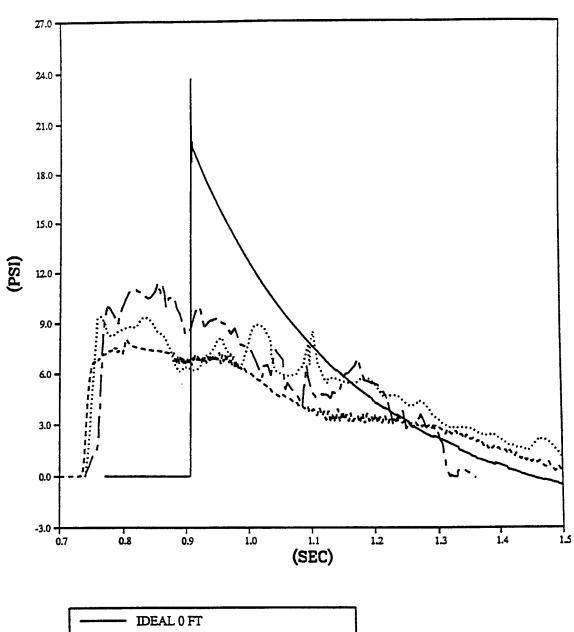


PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 2250 FEET (686 M)



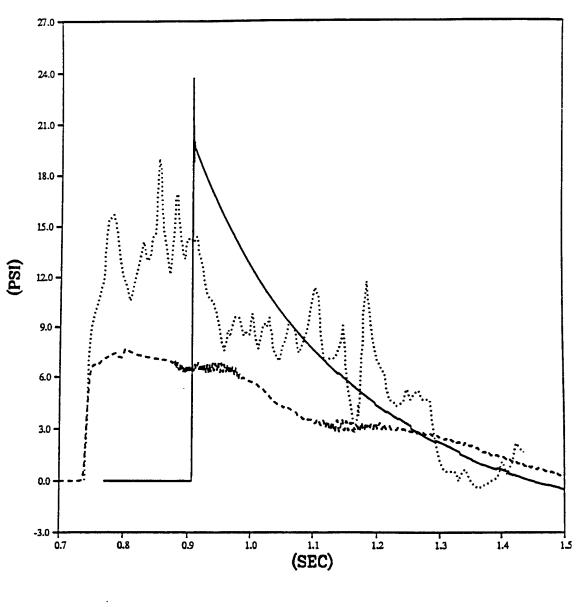
DEAL
DESERT CALC TL REV 6 JUL
BRL - 0 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 2500 FEET (762 M)



DEAL 0 FT
DESERT CALC 0FT TL REV 6JUL
BRL - 0 FT
SRI - 0 FT

PRISCILLA CALCULATION - DATA COMPARISONS OVERPRESSURE AT 2500 FEET (762 M)

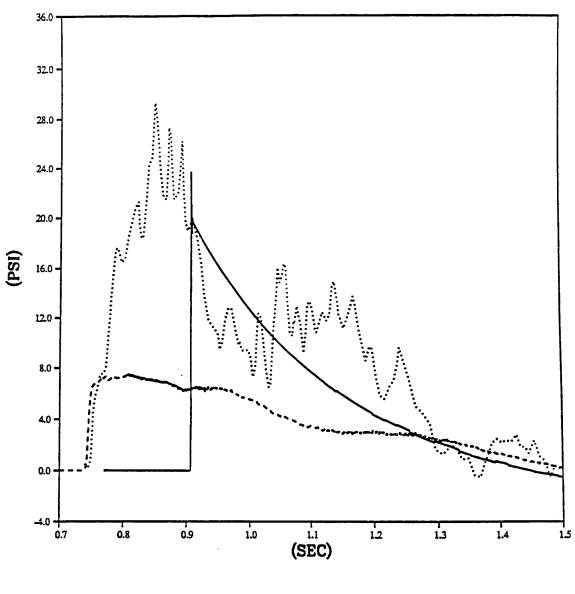


DEAL 3 FT

DESERT CALC 3 FT TL REV 6JUL

SRI - 3 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 2500 FEET (762 M)



DEAL 10 FT

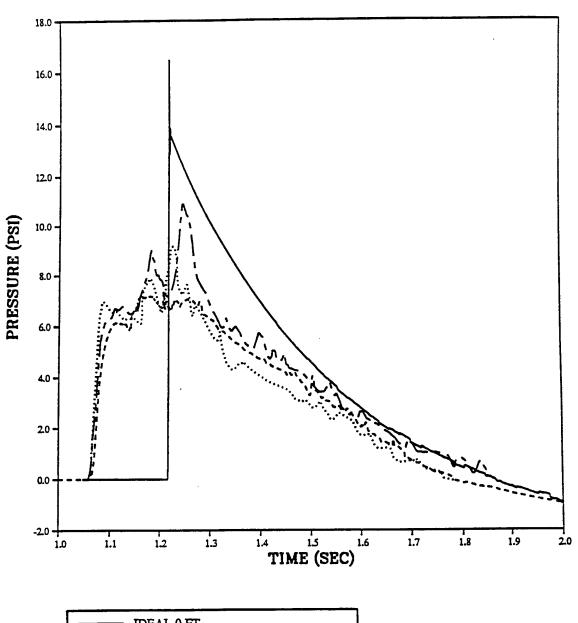
DESERT CALC 10 FT TL REV 6JUL

SRI - 10 FT

PRISCILLA

CALCULATION - DATA COMPARISONS

OVERPRESSURE AT 3000 FEET (914 M)

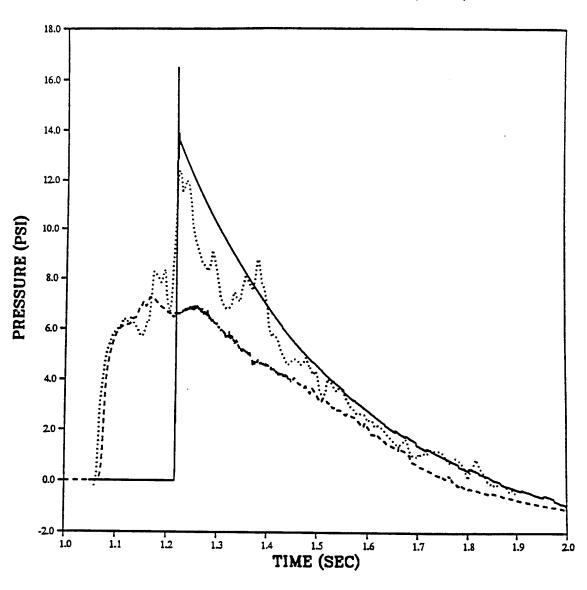


DEAL 0 FT
DESERT CALC 0 FT TL REV 6JUL

PRISCILLA

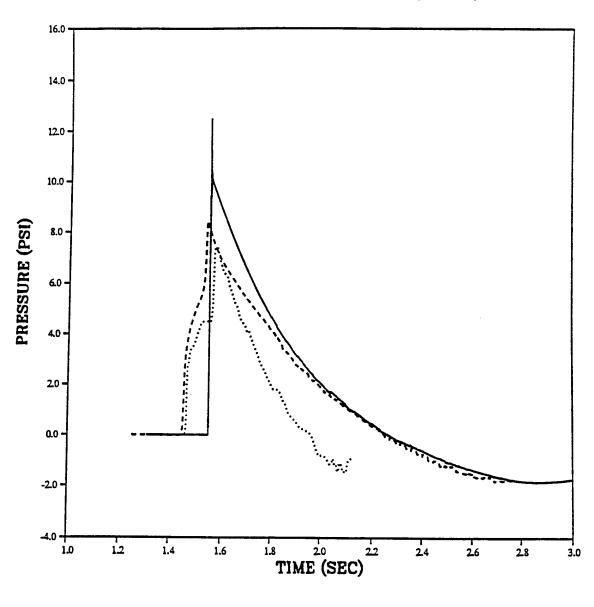
CALCULATION - DATA COMPARISONS

OVERPRESSURE AT 3000 FEET (914 M)



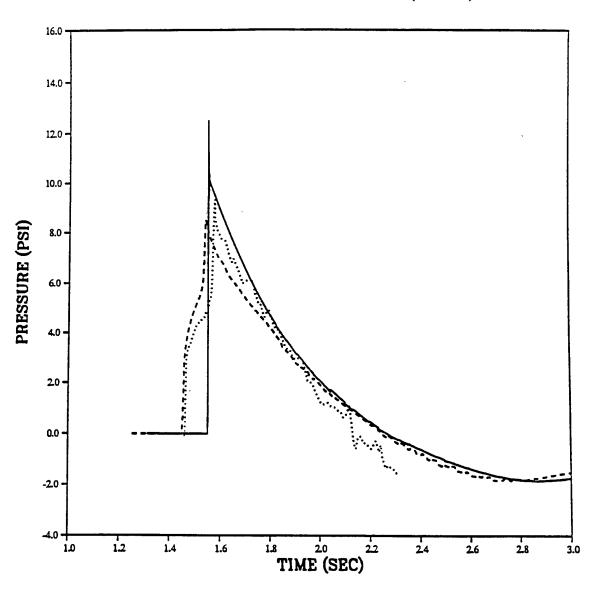
DEAL 10 FT
DESERT CALC 10 FT TL REV 6JUL
SRI 10 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 3500 FEET (1067 M)



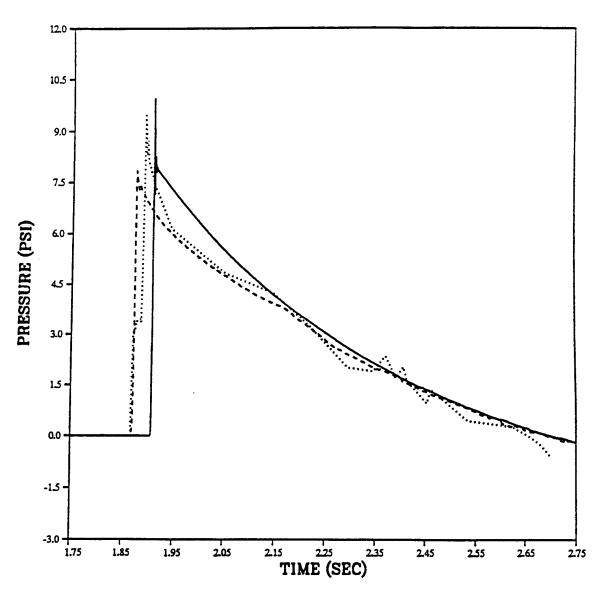
DEAL 0FT
DESERT CALC 0 FT TL REV 6JUL
SRI 0 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 3500 FEET (1067 M)



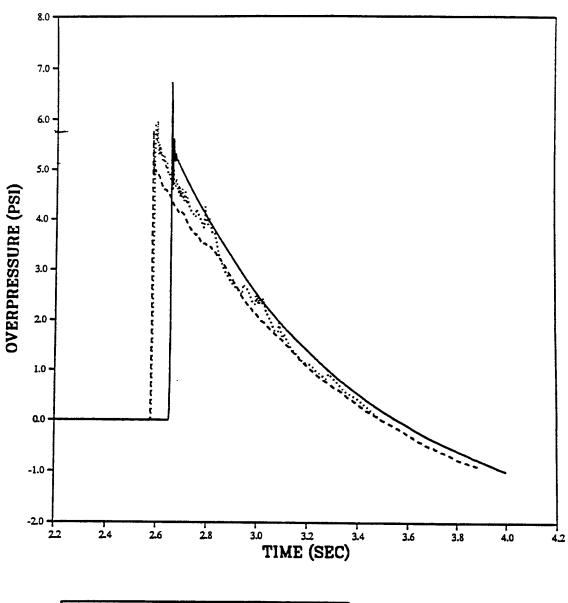
DEAL 3 FT
DESERT CALC 3FT TL REV 6JUL
SRI 3 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 4000 FEET (1219 M)



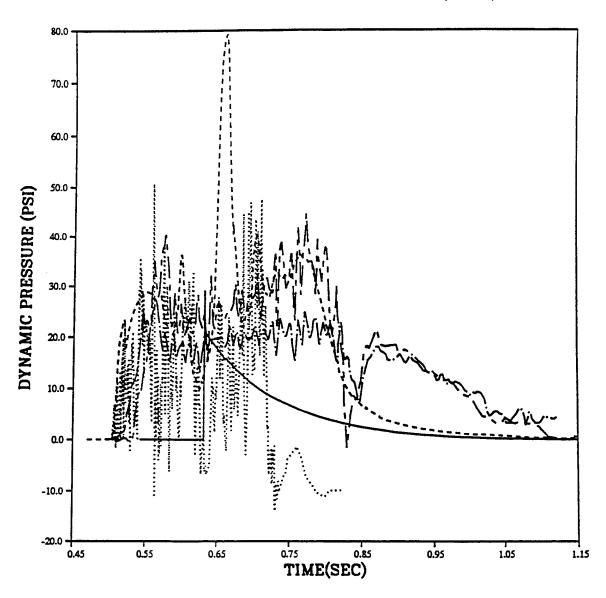
---- IDEAL 0 FT
---- DESERT CALC 0FT TL REV 6JUL
BRL 0 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
OVERPRESSURE AT 5000 FEET (1524 M)

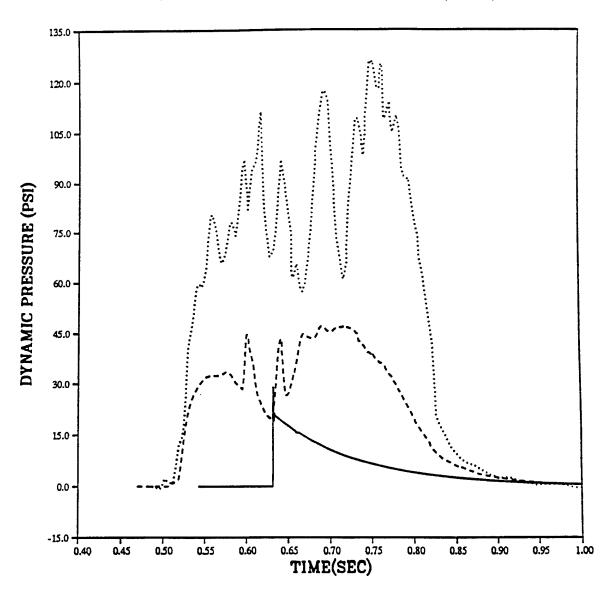


---- IDEAL 0 FT
---- DESERT CALC 3FT TL REV 6JUL
BRL - 0 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
DYNAMIC PRESSURE AT 2000 FEET (607 M)



PRISCILLA
CALCULATION - DATA COMPARISONS
DYNAMIC PRESSURE AT 2000 FEET (607 M)

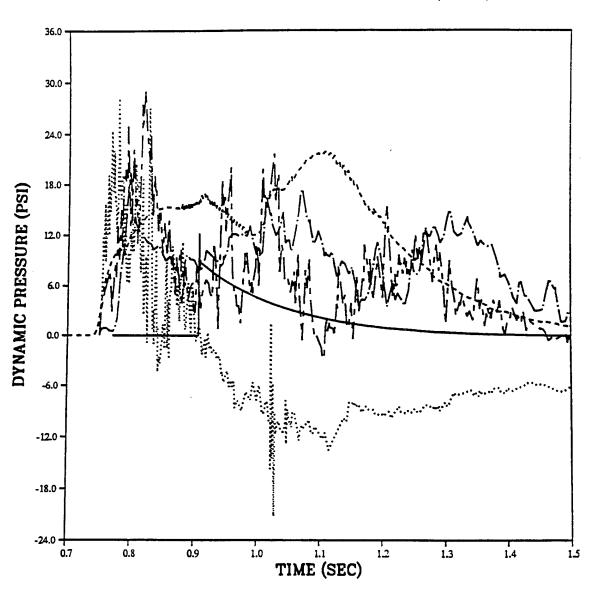


DEAL 10 FT
DESERT CALC 10 FT TL REV 6JUL
SRI - 10 FT

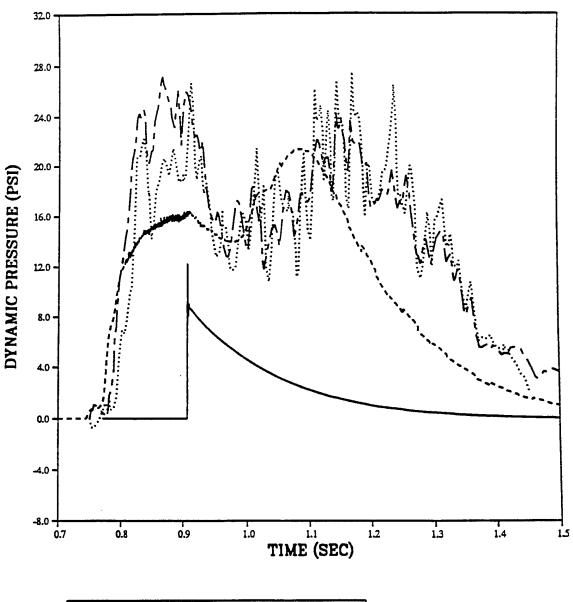
PRISCILLA

CALCULATION - DATA COMPARISONS

DYNAMIC PRESSURE AT 2500 FEET (762 M)



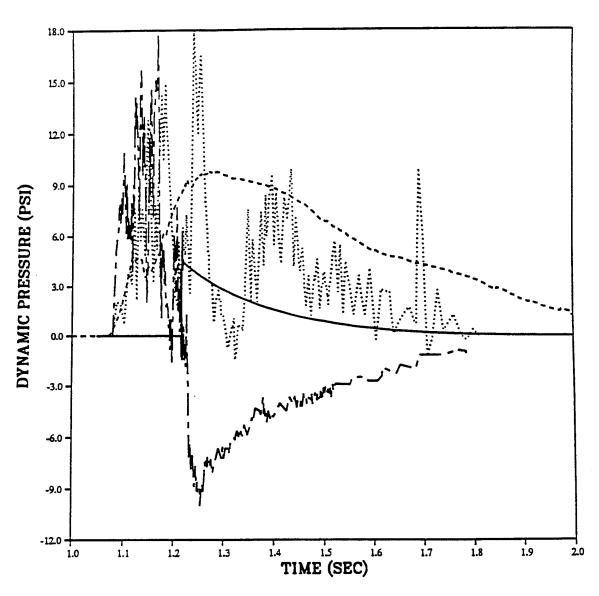
PRISCILLA
CALCULATION - DATA COMPARISONS
DYNAMIC PRESSURE AT 2500 FEET (762 M)



PRISCILLA

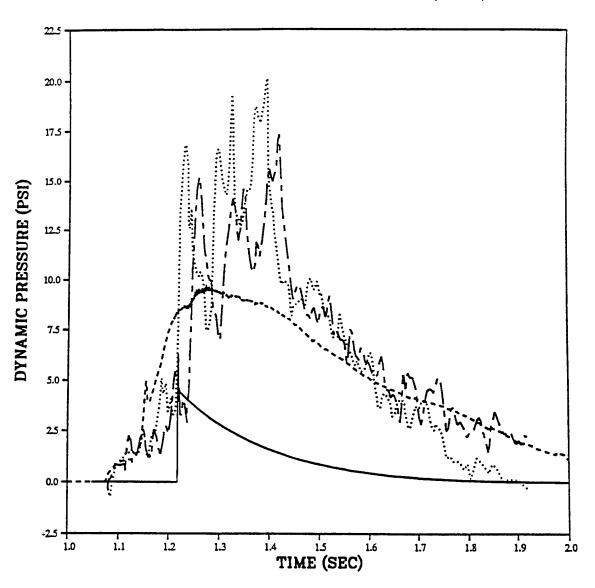
CALCULATION - DATA COMPARISONS

DYNAMIC PRESSURE AT 3000 FEET (762 M)

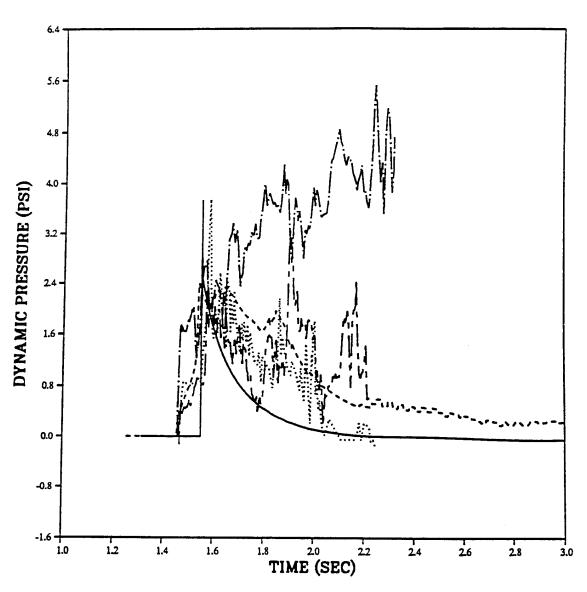


DEAL 3 FT
DESERT CALC 3 FT TL REV 6JUL
BRL - 3 FT
BRL - 3 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
DYNAMIC PRESSURE AT 3000 FEET (762 M)

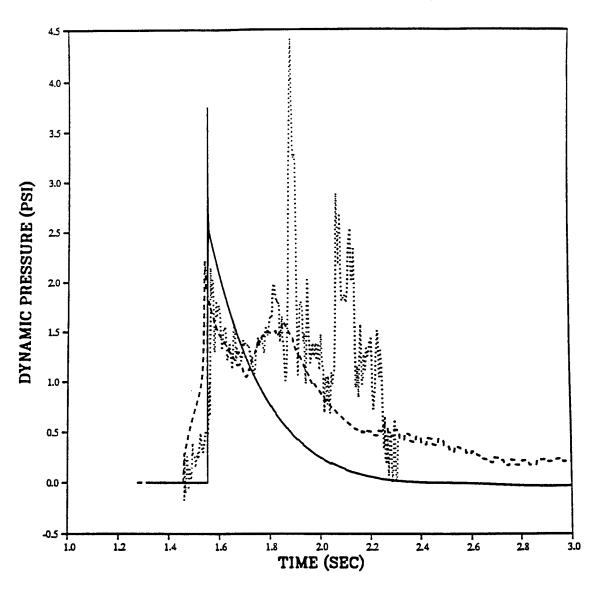


PRISCILLA
CALCULATION - DATA COMPARISONS
DYNAMIC PRESSURE AT 3500 FEET (1067 M)



---- IDEAL 3FT
---- DESERT CALC 3FT TL REV 6JUL
----- BRL - 3 FT
---- SRI - 3 FT
---- SRI - 3 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
DYNAMIC PRESSURE AT 3500 FEET (1067 M)

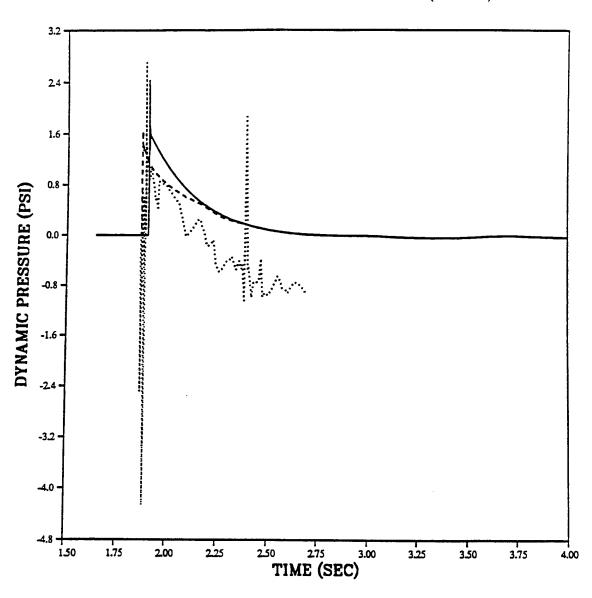


DEAL 10 FT

DESERT CALC 10 FT TL REV 6JUL

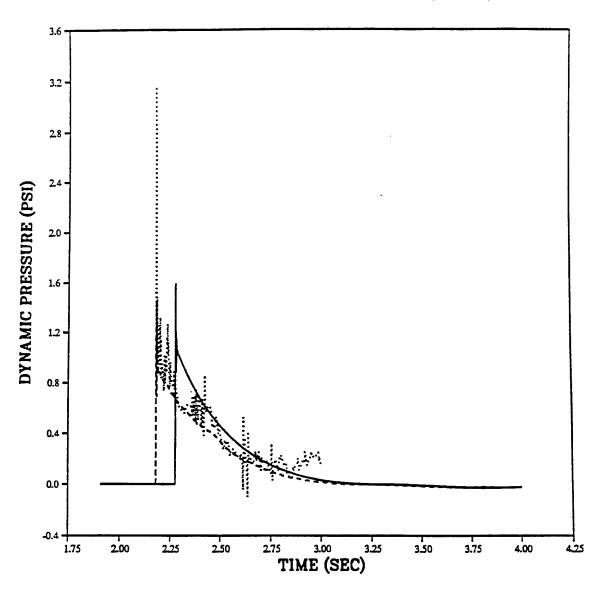
SRI - 10 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
DYNAMIC PRESSURE AT 4000 FEET (1219 M)



DEAL 3 FT
DESERT CALC 3FT TL REV 6JUL
BRL - 3 FT

PRISCILLA
CALCULATION - DATA COMPARISONS
DYNAMIC PRESSURE AT 4500 FEET (1372 M)



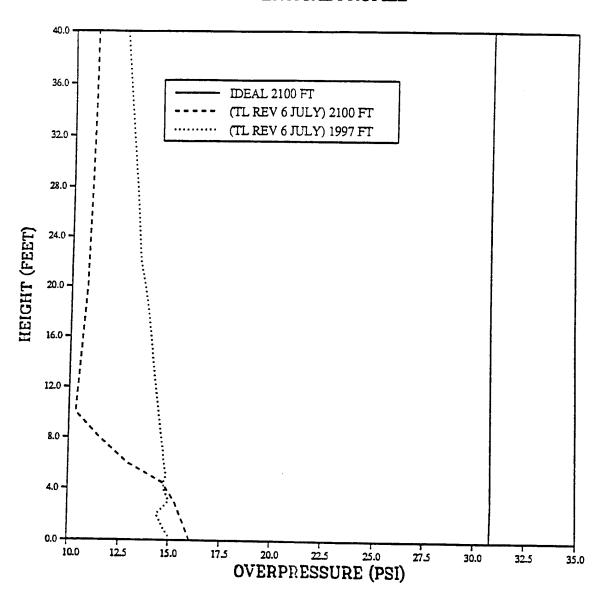
DEAL 3 FT
DESERT CALC 3FT TL REV 6JUL
BRL - 3 FT

INTENTIONALLY LEFT BLANK.

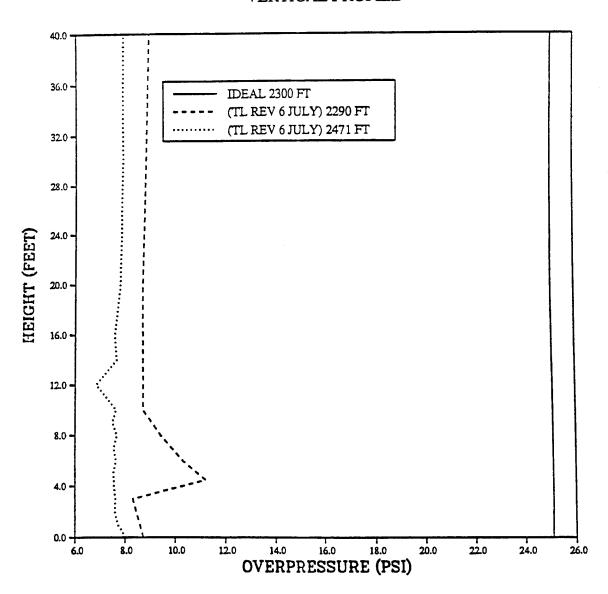
APPENDIX C HYDRODYNAMIC PARAMETERS AS A FUNCTION OF HEIGHT FOR SELECTED GROUND RANGES

This Appendix contains plots of important hydrodynamic parameters as a function of height above the surface at several ground ranges. The ground ranges were selected on the basis of predicted ideal overpressure levels. Results of calculated ideal and precursor parameters are displayed on each plot. Because of the problem with 32-bit truncation causing the precursor stations to move prior to shock arrival, most of the precursor results include two curves; one on either side of the ideal range.

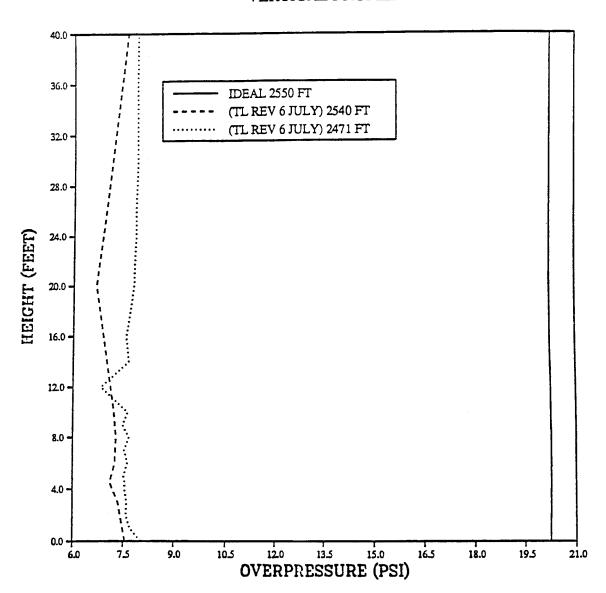
PRISCILLA DESERT OVERPRESSURE AT 2100 FEET VERTICAL PROFILE



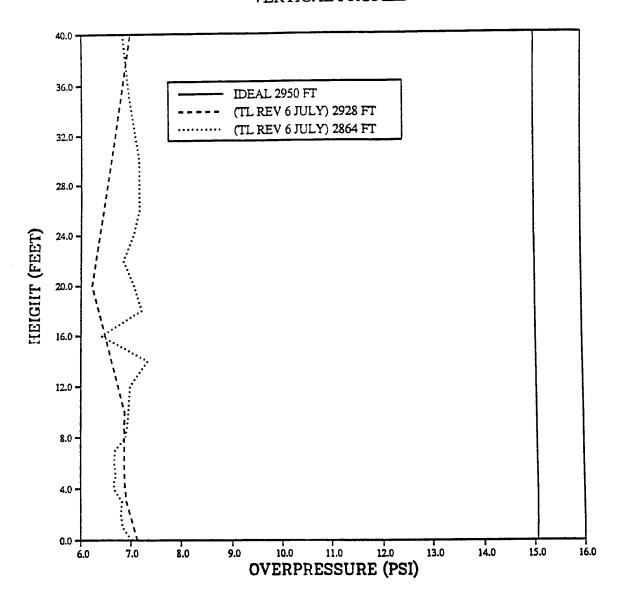
PRISCILLA DESERT OVERPRESSURE AT 2300 FEET VERTICAL PROFILE



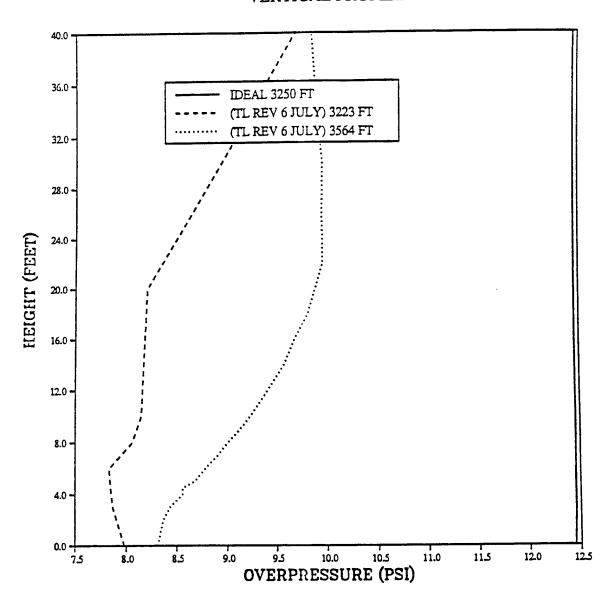
PRISCILLA DESERT OVERPRESSURE AT 2550 FEET VERTICAL PROFILE



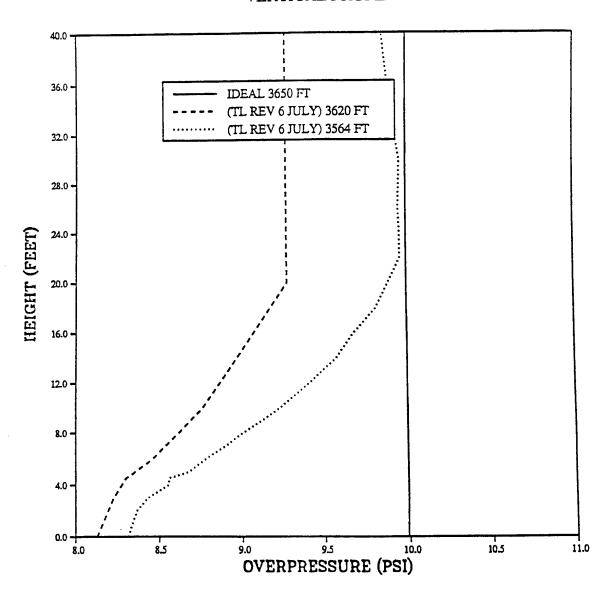
PRISCILLA DESERT OVERPRESSURE AT 2950 FEET VERTICAL PROFILE



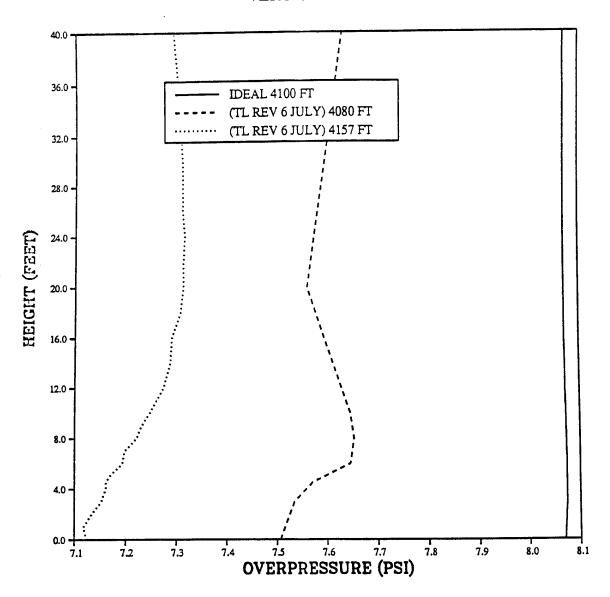
PRISCILLA DESERT OVERPRESSURE AT 3250 FEET VERTICAL PROFILE



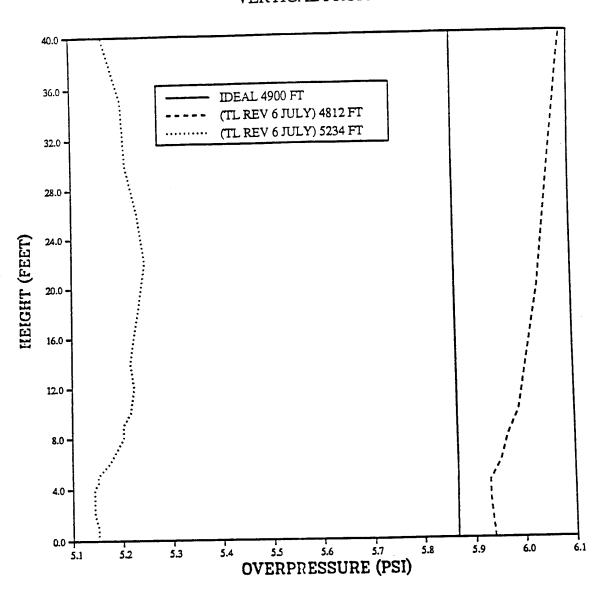
PRISCILLA DESERT OVERPRESSURE AT 3650 FEET VERTICAL PROFILE



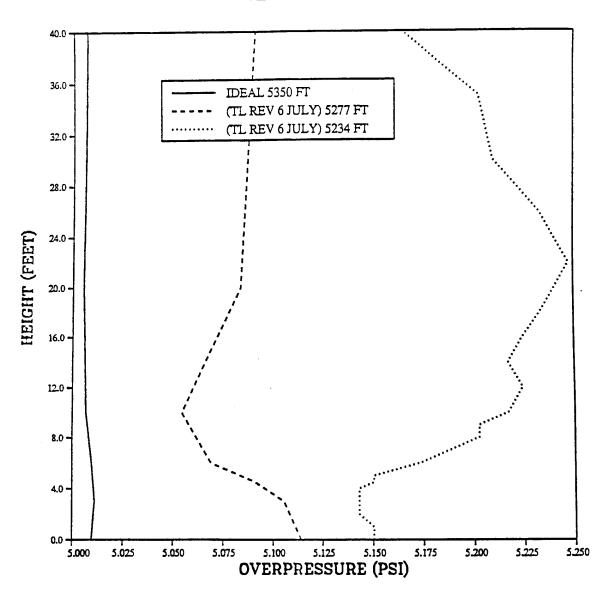
PRISCILLA DESERT OVERPRESSURE AT 4100 FEET VERTICAL PROFILE



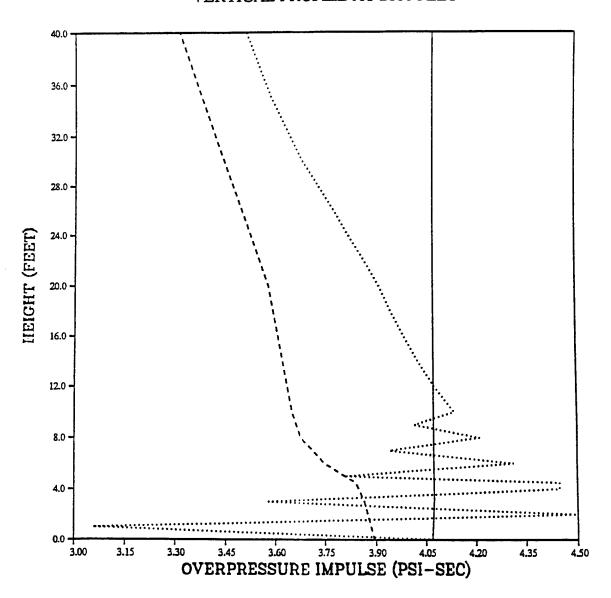
PRISCILLA DESERT OVERPRESSURE AT 4900 FEET VERTICAL PROFILE



PRISCILLA DESERT OVERPRESSURE AT 5350 FEET VERTICAL PROFILE

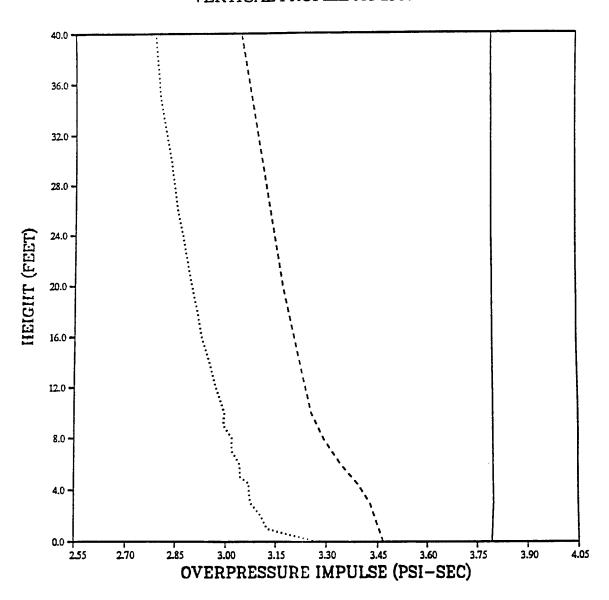


PRISCILLA DESERT OVERPRESSURE IMPULSE VERTICAL PROFILE AT 2100 FEET



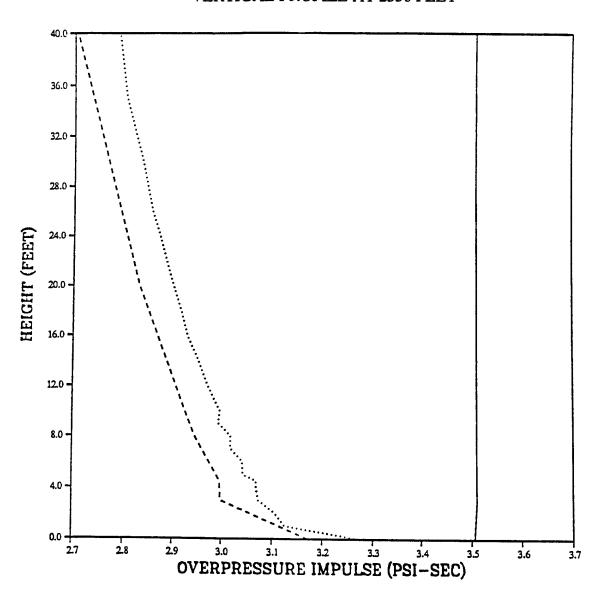
—— IDEAL 2100 FT ---- 2100 FT (TL REV 6 JULY) ----- 1997 FT (TL REV 6 JULY)

PRISCILLA DESERT OVERPRESSURE IMPULSE VERTICAL PROFILE AT 2300 FEET



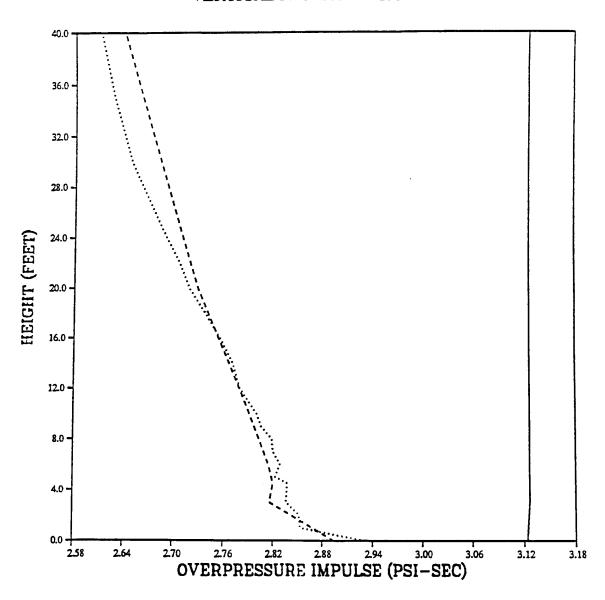
DEAL 2300 FT ---- 2290 FT (TL REV 6 JULY)
2471 FT (TL REV 6 JULY)

PRISCILLA DESERT OVERPRESSURE IMPULSE VERTICAL PROFILE AT 2550 FEET



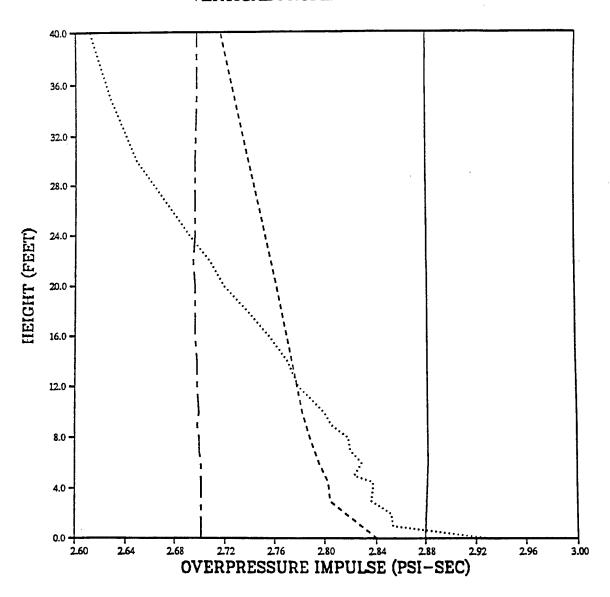
---- IDEAL 2550 FT
---- 2540 FT (TL REV 6 JULY)
----- 2471 FT (TL REV 6 JULY)

PRISCILLA DESERT OVERPRESSURE IMPULSE VERTICAL PROFILE AT 2950 FEET



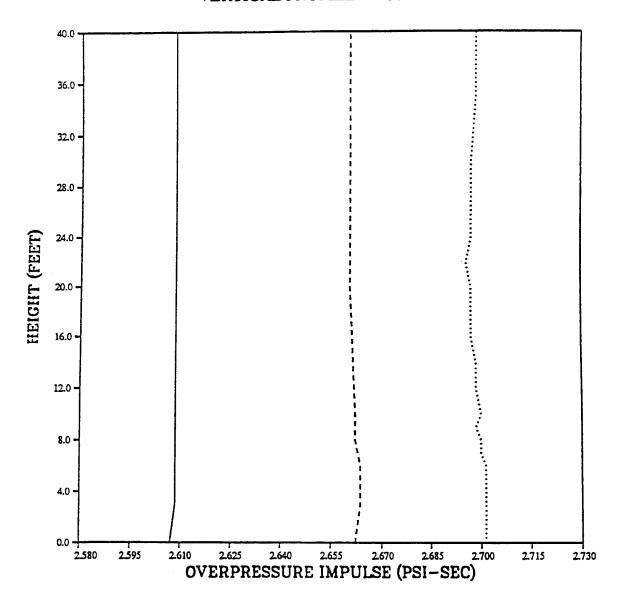
---- IDEAL 2950 FT
---- 2928 FT (TL REV 6 JULY)
----- 2864 FT (TL REV 6 JULY)

PRISCILLA DESERT OVERPRESSURE IMPULSE VERTICAL PROFILE AT 3250 FEET



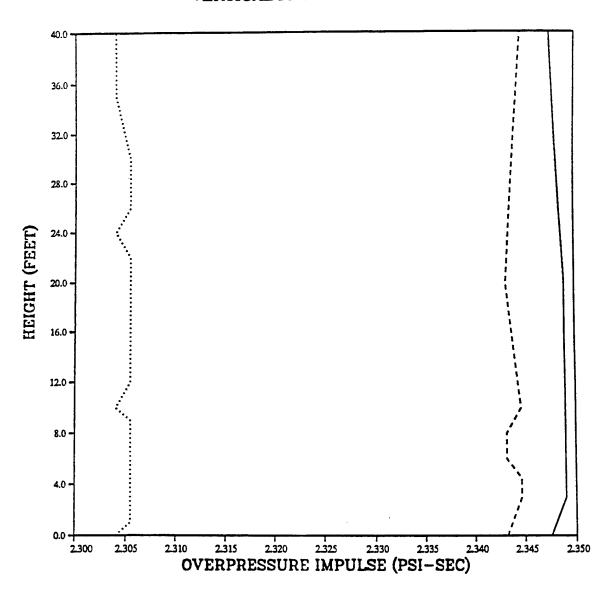
----- IDEAL 3250 FT
----- 3223 FT (TL REV 6 JULY)
----- 2864 FT (TL REV 6 JULY)
---- 3564 FT (TL REV 6 JULY)

PRISCILLA DESERT OVERPRESSURE IMPULSE VERTICAL PROFILE AT 3650 FEET



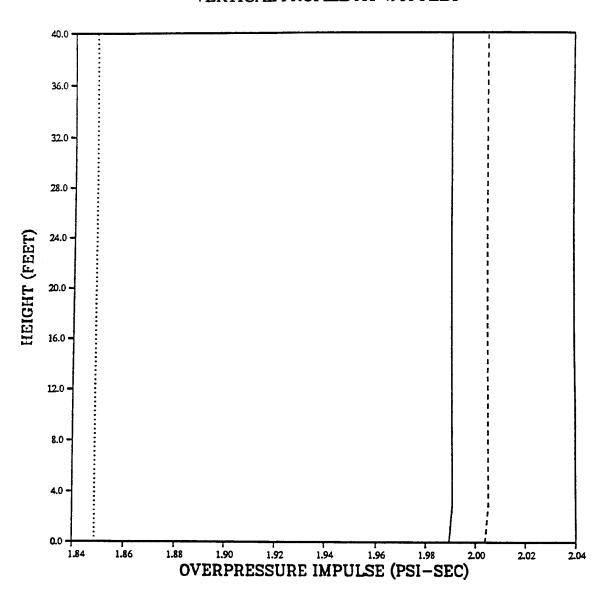
----- IDEAL 3650 FT ---- 3620 FT (TL REV 6 JULY) ------ 3564 FT (TL REV 6 JULY)

PRISCILLA DESERT OVERPRESSURE IMPULSE VERTICAL PROFILE AT 4100 FEET



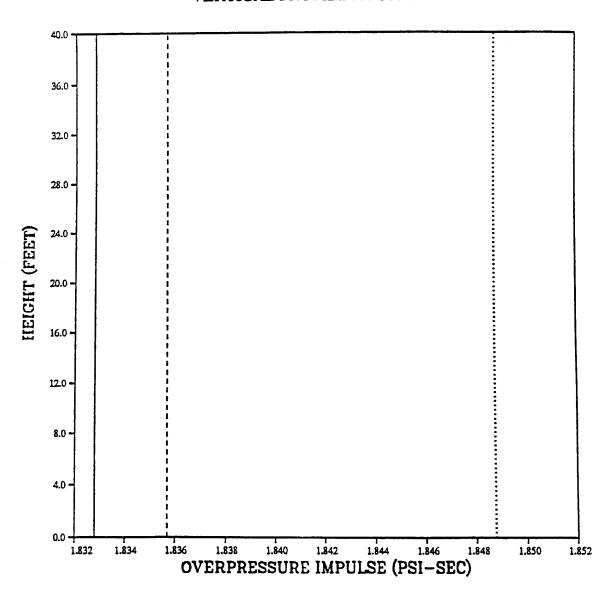
---- IDEAL 4100 FT ---- 4080 FT (TL REV 6 JULY) ------ 4157 FT (TL REV 6 JULY)

PRISCILLA DESERT OVERPRESSURE IMPULSE VERTICAL PROFILE AT 4900 FEET



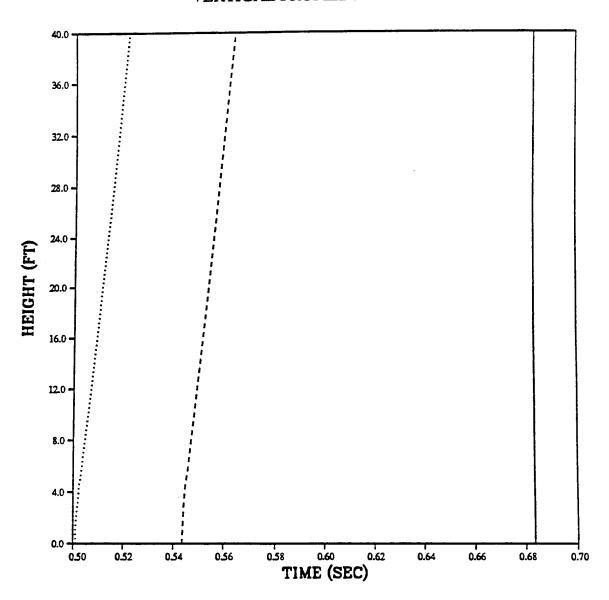
---- IDEAL 4900 FT
---- 4812 FT (TL REV 6 JULY)
5234 FT (TL REV 6 JULY)

PRISCILLA DESERT OVERPRESSURE IMPULSE VERTICAL PROFILE AT 5350 FEET



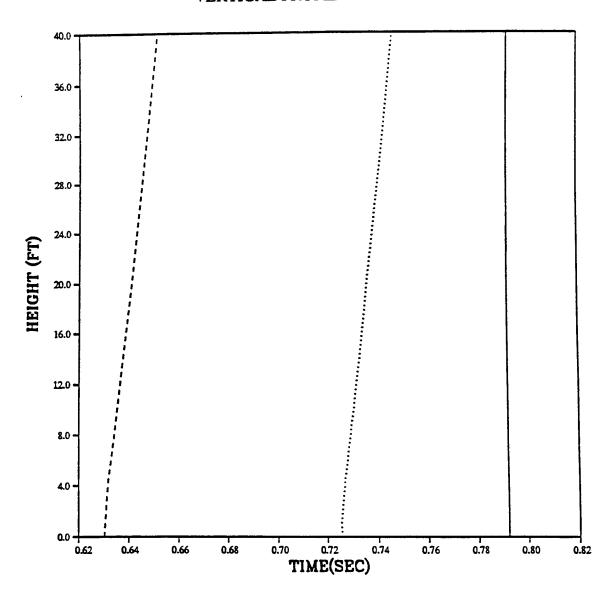
---- IDEAL 5350 FT ---- 5277 FT (TL REV 6 JULY) ----- 5234 FT (TL REV 6 JULY)

PRISCILLA ARRIVAL TIME VERTICAL PROFILE AT 2100 FEET



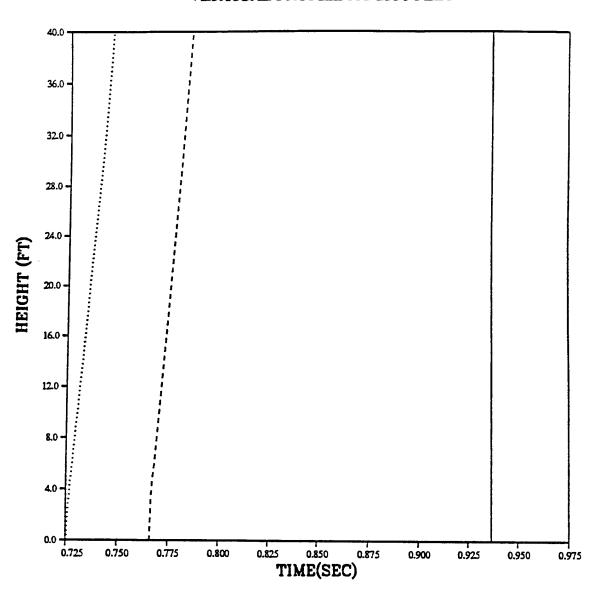
---- IDEAL 2100 FT
---- (TL REV 6 JULY) 2100 FT
..... (TL REV 6 JULY) 1997 FT

PRISCILLA ARRIVAL TIME VERTICAL PROFILE AT 2300 FEET



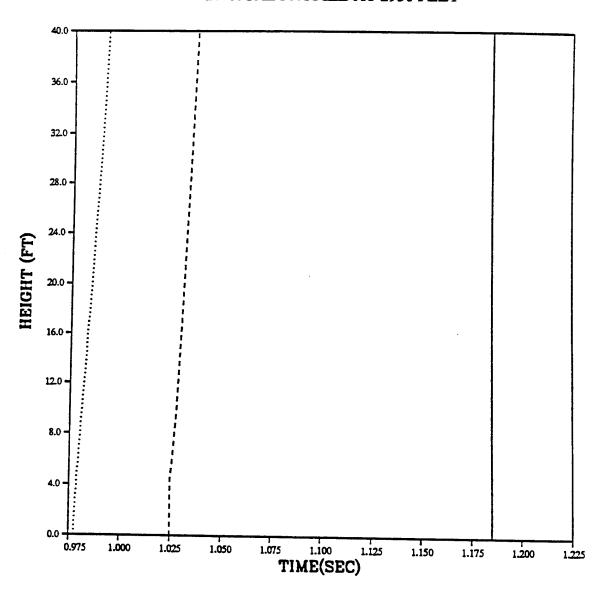
---- IDEAL
---- (TL REV 6 JULY) 2290 FT
----- (TL REV 6 JULY) 2471 FT

PRISCILLA ARRIVAL TIME VERTICAL PROFILE AT 2550 FEET



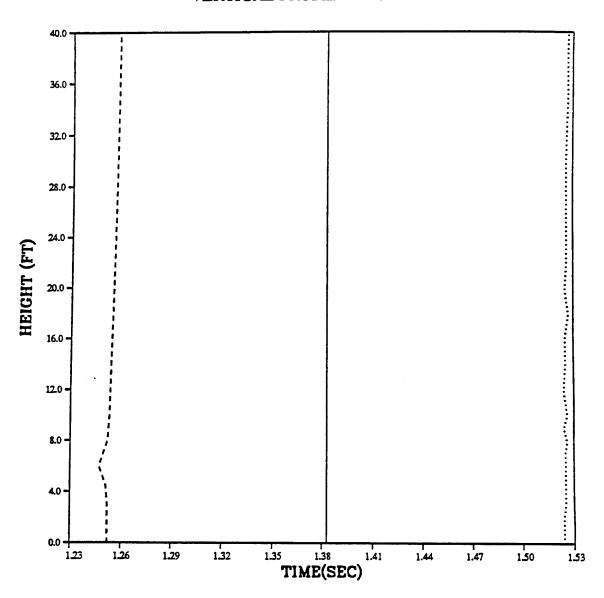
----- IDEAL 2550 FT
----- (TL REV 6 JULY) 2550 FT
----- (TL REV 6 JULY) 2471 FT

PRISCILLA ARRIVAL TIME VERTICAL PROFILE AT 2950 FEET



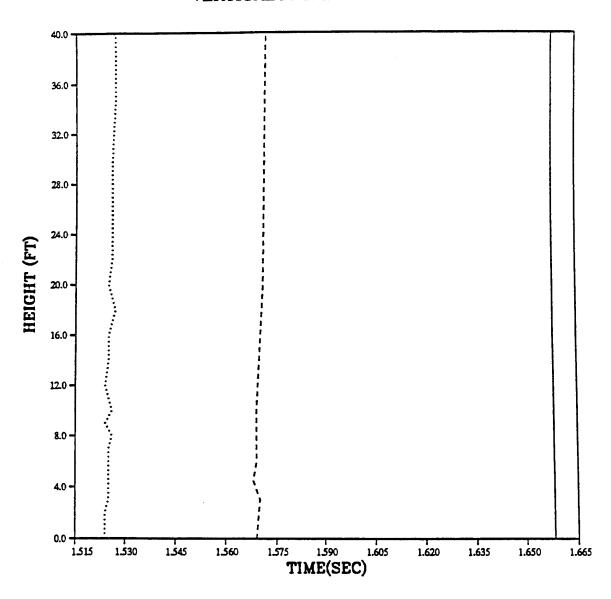
---- IDEAL 2950 FT
---- (TL REV 6 JULY) 2928 FT
---- (TL REV 6 JULY) 2864 FT

PRISCILLA ARRIVAL TIME VERTICAL PROFILE AT 3250 FEET



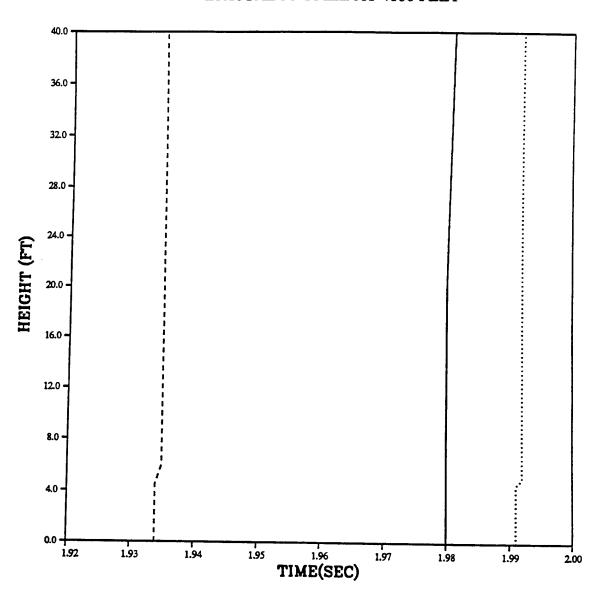
---- IDEAL 3250 FT
---- (TL REV 6 JULY) 3223 FT
..... (TL REV 6 JULY) 3564 FT

PRISCILLA ARRIVAL TIME VERTICAL PROFILE AT 3650 FEET



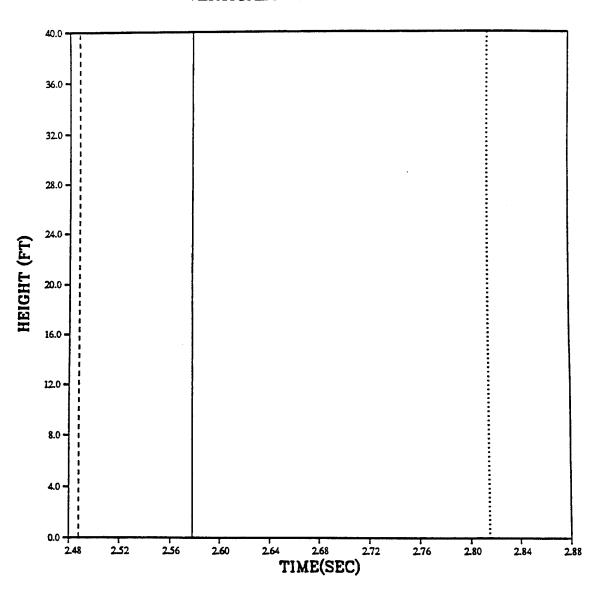
---- IDEAL 3650 FT
---- (TL REV 6 JULY) 3620 FT
---- (TL REV 6 JULY) 3564 FT

PRISCILLA ARRIVAL TIME VERTICAL PROFILE AT 4100 FEET



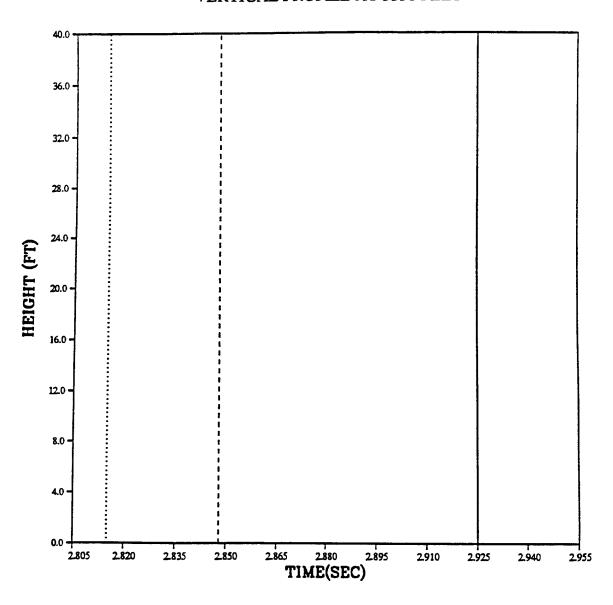
----- IDEAL 4100 FT
----- (TL REV 6 JULY) 4080 FT
----- (TL REV 6 JULY) 4157 FT

PRISCILLA ARRIVAL TIME VERTICAL PROFILE AT 4900 FEET



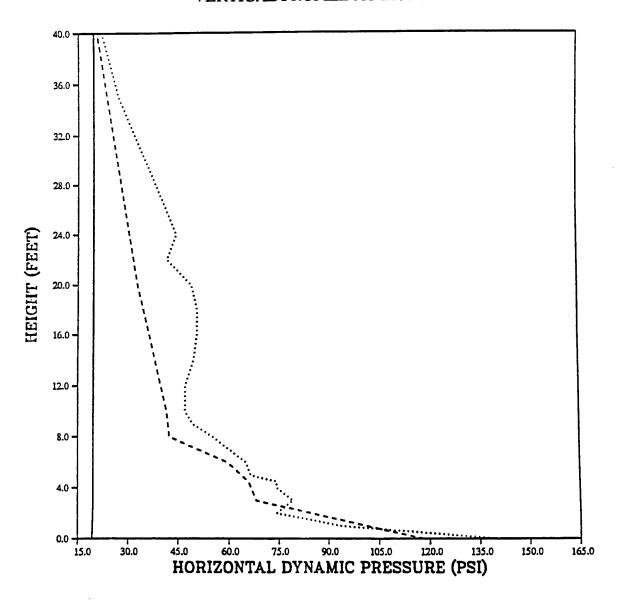
---- IDEAL 4900 FT
---- (TL REV 6 JULY) 4812 FT
----- (TL REV 6 JULY) 5234 FT

PRISCILLA ARRIVAL TIME VERTICAL PROFILE AT 5350 FEET



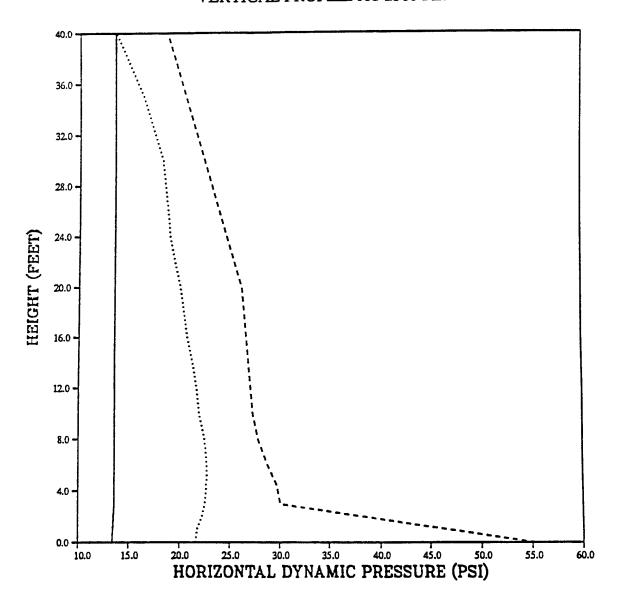
---- IDEAL 5350 FT
---- (TL REV 6 JULY) 5227 FT
----- (TL REV 6 JULY) 5234 FT

PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS VERTICAL PROFILE AT 2100 FEET



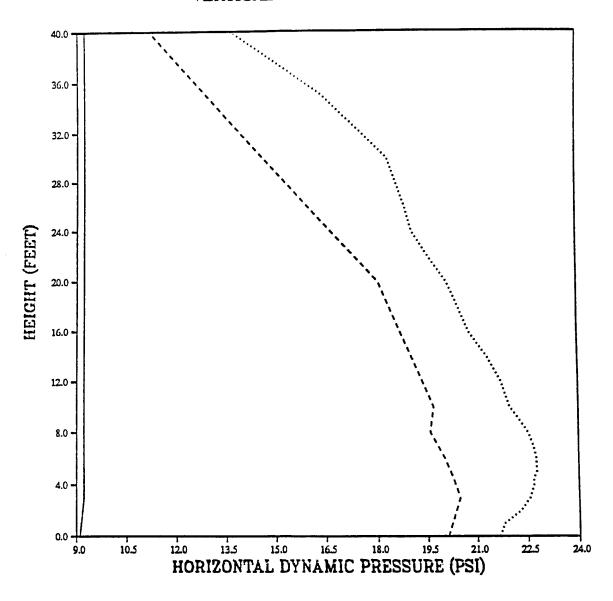
DEAL 2100 FT
THERMAL LAYER 2100 FT (REV 6 JULY)
THERMAL LAYER 1997 FT (REV 6 JULY)

PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS VERTICAL PROFILE AT 2300 FEET



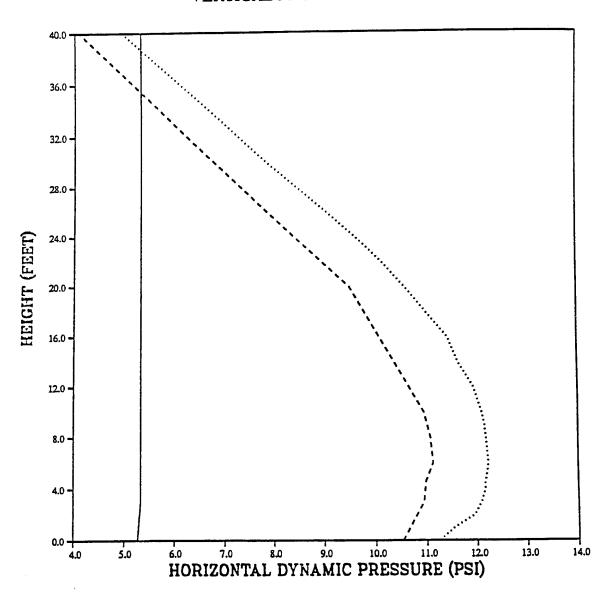
—— IDEAL 2300 FT ---- 2290 FT (TL REV 6 JULY) 2471 FT (TL REV 6 JULY)

PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS VERTICAL PROFILE AT 2550 FEET

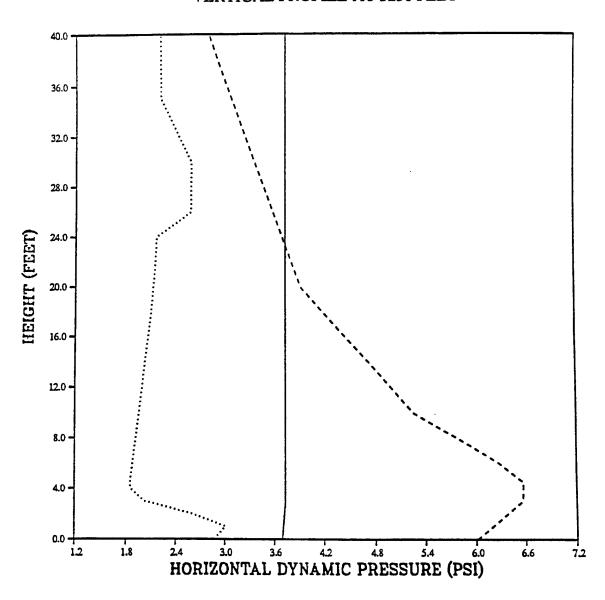


---- IDEAL 2550 FT
---- 2540 FT (TL REV 6 JULY)
----- 2471 FT (TL REV 6 JULY)

PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS VERTICAL PROFILE AT 2950 FEET

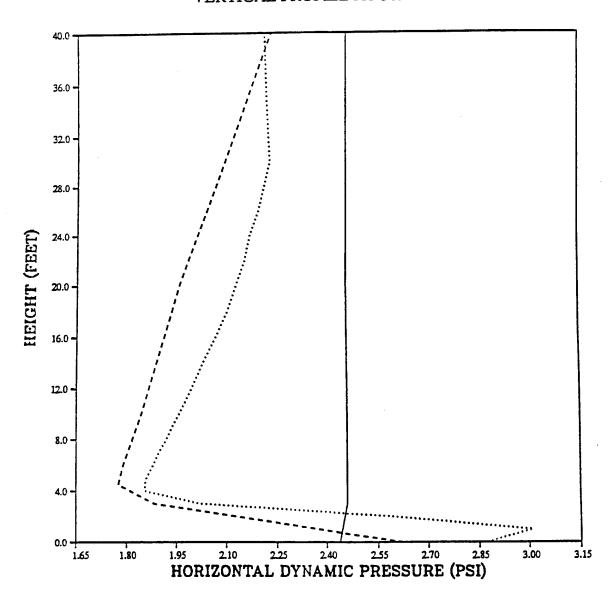


PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS VERTICAL PROFILE AT 3250 FEET



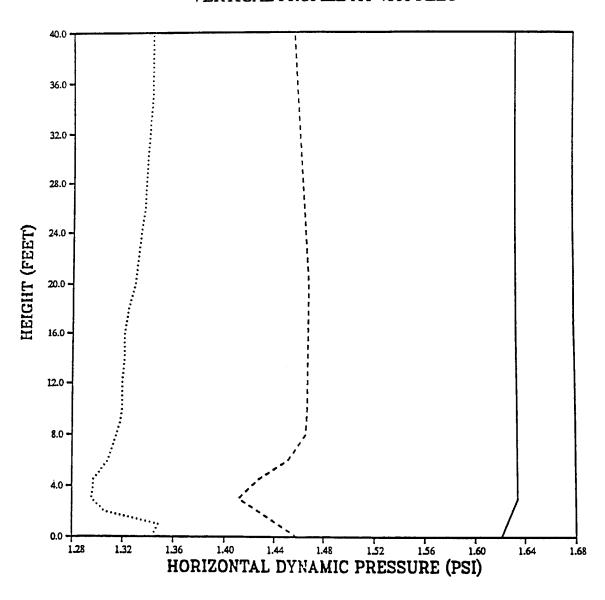
---- IDEAL 3250 FT
---- 3223 FT (TL REV 6 JULY)
----- 3564 FT (TL REV 6 JULY)

PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS VERTICAL PROFILE AT 3650 FEET



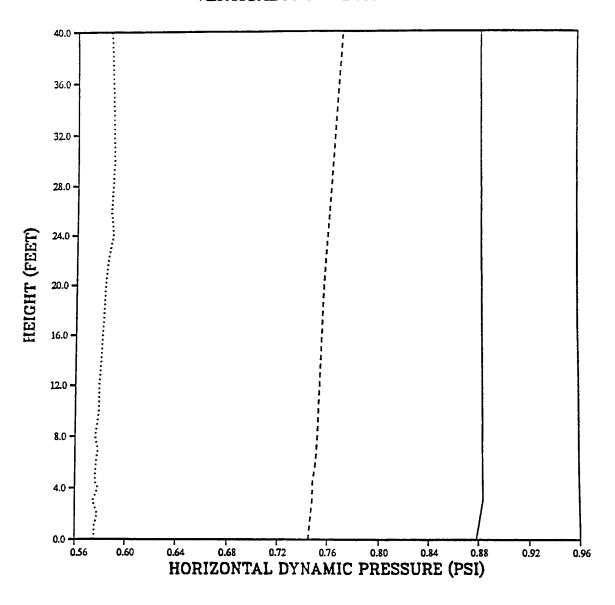
---- IDEAL 3650 FT
---- 3620 FT (TL REV 6 JULY)
----- 3564 FT (TL REV 6 JULY)

PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS VERTICAL PROFILE AT 4100 FEET



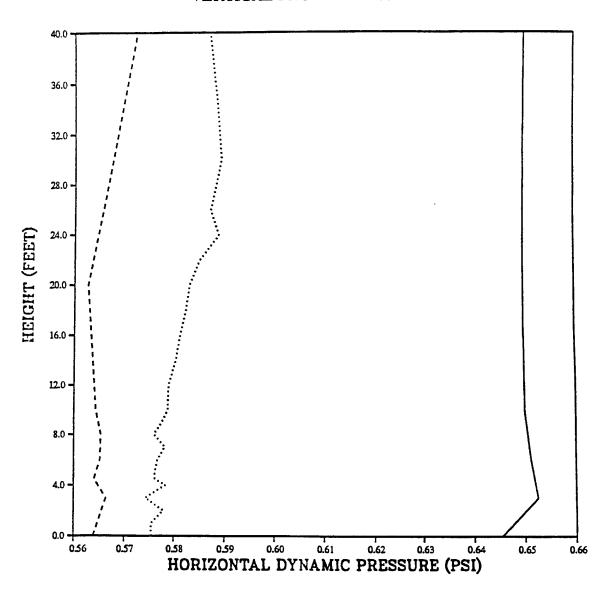
---- IDEAL 4100 FT ---- 4080 FT (TL REV 6 JULY) ----- 4157 FT (TL REV 6 JULY)

PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS VERTICAL PROFILE AT 4900 FEET



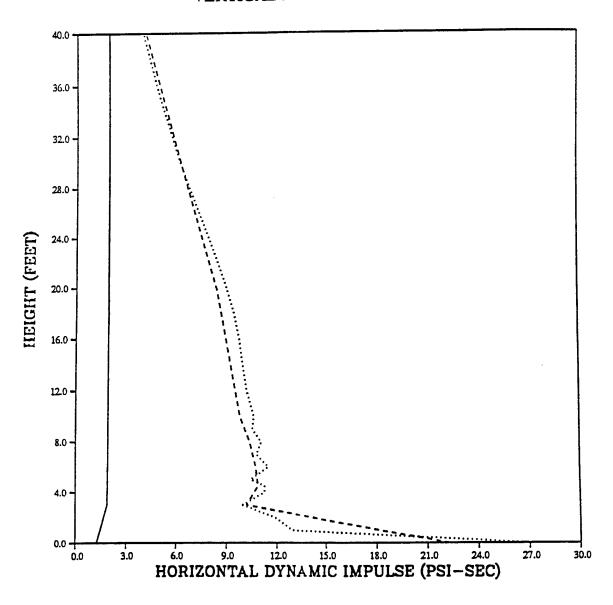
----- IDEAL 4900 FT ----- 4812 FT (TL REV 6 JULY) ------ 5234 FT (TL REV 6 JULY)

PRISCILLA DESERT HORIZONTAL DYNAMIC PRESSURE PEAKS VERTICAL PROFILE AT 5350 FEET



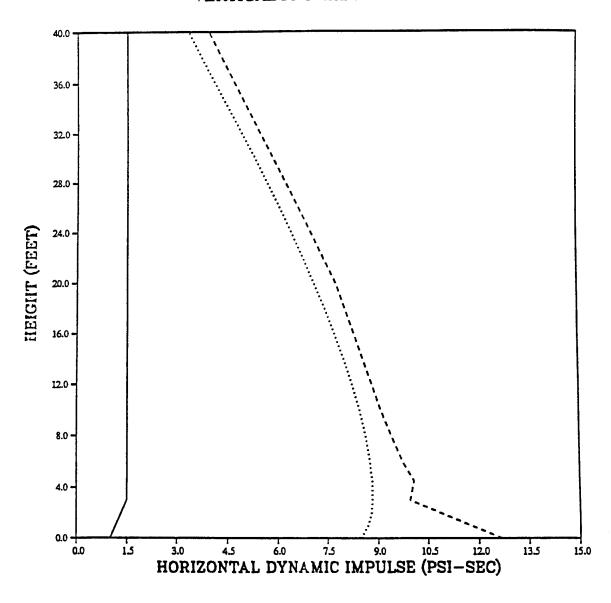
DEAL 5350 FT
---- 5277 FT (TL REV 6 JULY)
5234 FT (TL REV 6 JULY)

DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE VERTICAL PROFILE AT 2100 FEET



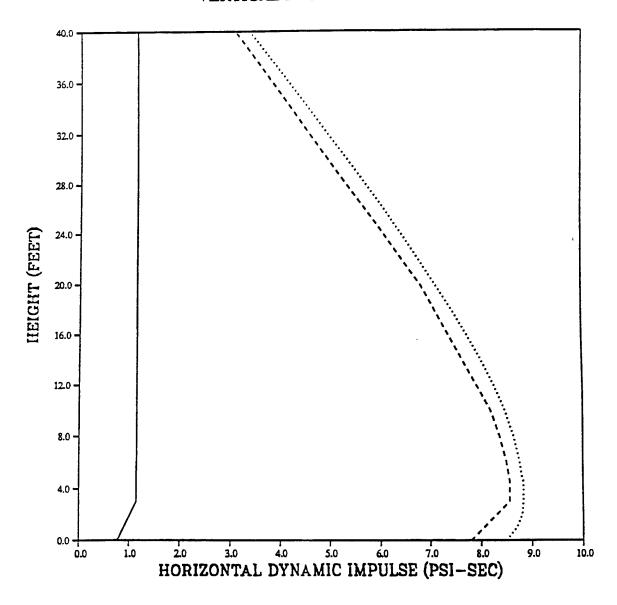
---- IDEAL 2100 FT ---- 2100 FT (TL REV 6 JULY) ----- 1997 FT (TL REV 6 JULY)

DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE VERTICAL PROFILE AT 2300 FEET



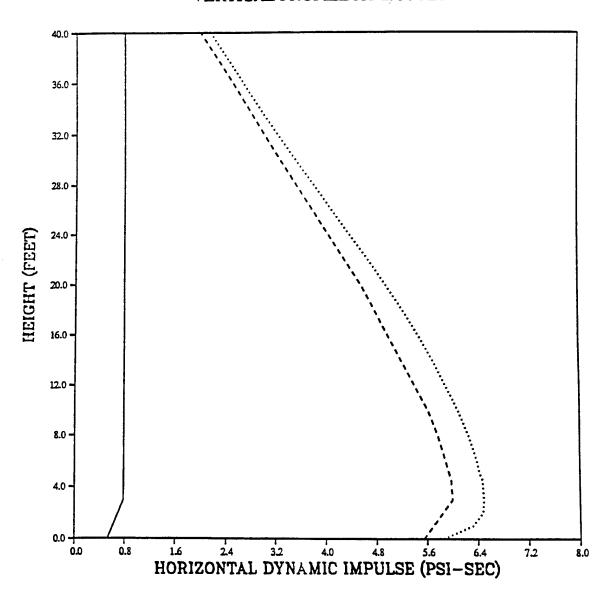
----- IDEAL 2300 FT ----- 2290 FT (TL REV 6 JULY) ------ 2471 FT (TL REV 6 JULY)

DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE VERTICAL PROFILE AT 2550 FEET



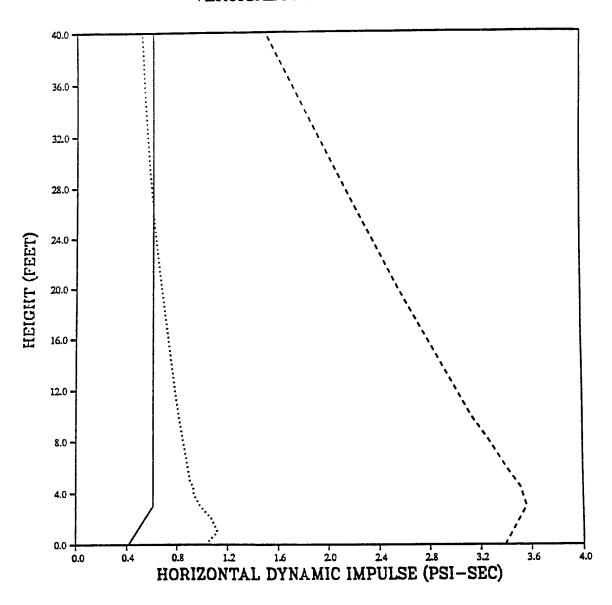
---- IDEAL 2550 FT
---- 2540 FT (TL REV 6 JULY)
----- 2471 FT (TL REV 6 JULY)

DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE VERTICAL PROFILE AT 2950 FEET



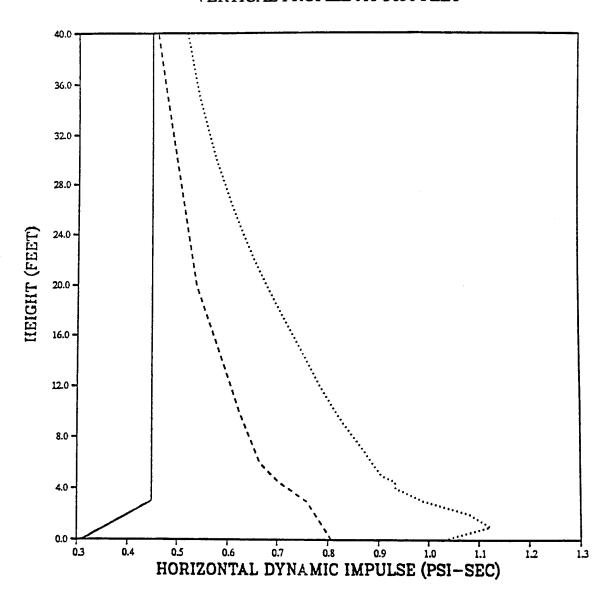
---- IDEAL 2950 FT
---- 2928 FT (TL REV 6 JULY)
----- 2864 FT (TL REV 6 JULY)

DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE VERTICAL PROFILE AT 3250 FEET



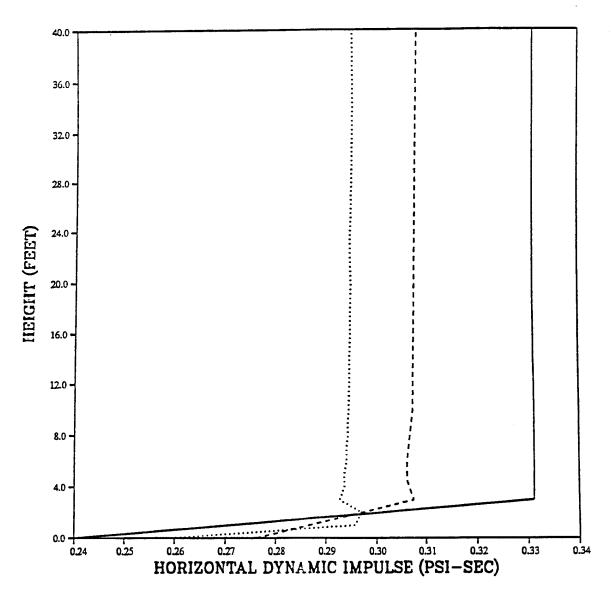
---- IDEAL 3250 FT ---- 3223 FT (TL REV 6 JULY) ----- 3564 FT (TL REV 6 JULY)

DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE VERTICAL PROFILE AT 3650 FEET



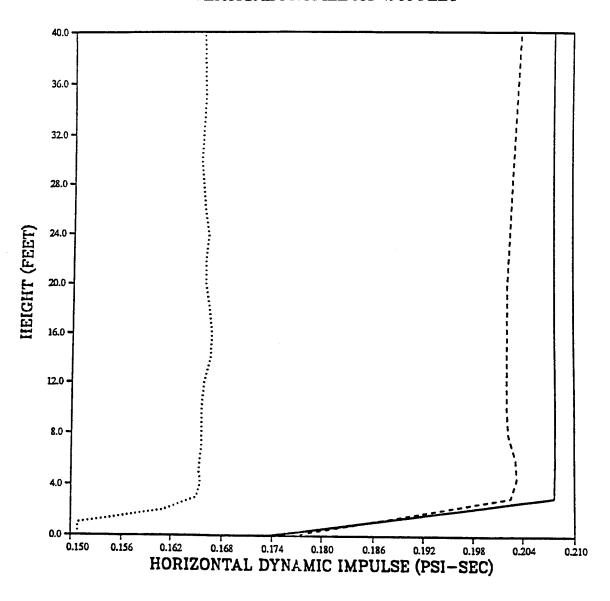
---- IDEAL 3650 FT ---- 3620 FT (TL REV 6 JULY) ----- 3564 FT (TL REV 6 JULY)

DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE VERTICAL PROFILE AT 4100 FEET



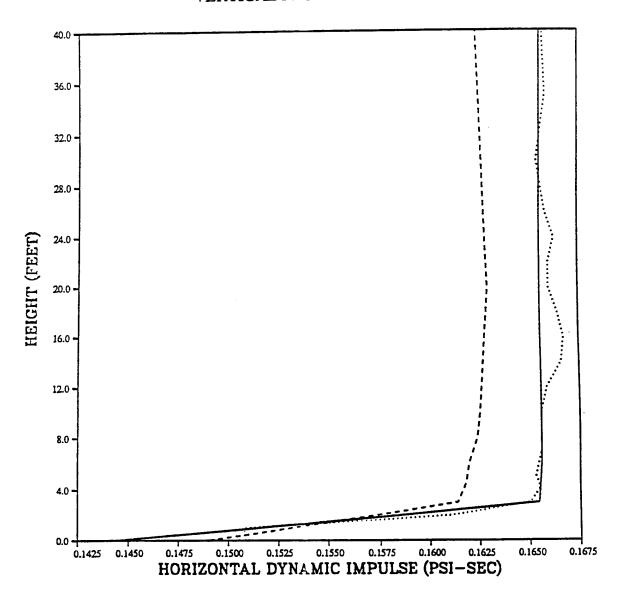
----- IDEAL 4100 FT ----- 4080 FT (TL REV 6 JULY) ------ 4157 FT (TL REV 6 JULY)

DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE VERTICAL PROFILE AT 4900 FEET



----- IDEAL 4900 FT
----- 4812 FT (TL REV 6 JULY)
------ 5234 FT (TL REV 6 JULY)

DESERT PRISCILLA HORIZONTAL DYNAMIC PRESSURE IMPULSE VERTICAL PROFILE AT 5350 FEET



---- IDEAL 5350 FT ---- 5277 FT (TL REV 6 JULY) ----- 5234 FT (TL REV 6 JULY)

APPENDIX D

CONVERSION TABLE

Conversion factors for U.S. Customary to metric (SI) units of measurement

➤ BY -

MULTIPLY -

TO GET

MOLITELI —	—— BY ◀	DIVIDE
TO GET ◀		1
angstrom	1.000 000 X E -10	meters (m)
atmosphere (normal)	1.013 25 X E +2	kilo pascal (kPa)
bar	1.000 000 X E +2	kilo pascal (kPa)
barn	1.000 000 X E -28	meter ² (m ²)
British thermal unit (thermochemical)	1.054 350 X E +3	joule (J)
calorie (thermochemical)	4.184 000	joule (J)
cal (thermochemical)/cm ²	4.184 000 X E -2	mega joule/m ² (MJ/m ²)
curie	3.700 000 X E +1	* giga becquerel (GBq)
degree (angle)	1.745 329 X E -2	radian (rad)
degree Fahrenheit	$t_{\kappa} = (t \cdot f + 459.67)/1.8$	degree kelvin (K)
electron volt	1.602 19 X E -19	joule (J)
erg	1.000 000 X E -7	joule (J)
erg/second	1.000 000 X E -7	watt (W)
foot	3.048 000 X E -1	meter (m)
foot-pound-force	1.355 818	joule (J)
gallon (U.S. liquid)	3.785 412 X E -3	meter 3 (m ³)
inch	2.540 000 X E -2	meter (m)
jerk	1.000 000 X E +9	joule (J)
joule/kilogram (J/kg) (radiation dose		
absorbed)	1.000 000	Gray (Gy)
kilotons	4.183	terajoules
kip (1000 lbf)	4.448 222 X E +3	newton (N)
kip/inch ² (ksi)	6.894 757 X E +3	kilo pascal (kPa)
ktap	0.05 1 70 7 11 2 15	newton-second/m ²
	1.000 000 X E +2	$(N-s/m^2)$
micron	1.000 000 X E -6	meter (m)
mil	2.540 000 X E -5	meter (m)
mile (international)	1.609 344 X E +3	meter (m)
ounce	2.834 952 X E -2	kilogram (kg)
pound-force (lbs avoirdupois)	4.448 222	newton (N)
•	1.129 848 X E -1	newton-meter (N•m)
pound-force inch		
pound-force/inch	1.751 268 X E +2	newton/meter (N/m)
pound-force/foot ²	4.788 026 X E -2	kilo pascal (kPa)
pound-force/inch ² (psi)	6.894 757	kilo pascal (kPa)
pound-mass (lbm avoirdupois)	4.535 924 X E -1	kilogram (kg)
pound-mass-foot ² (moment of intertia)		kilogram-meter ²
_	4.214 011 X E -2	(kg•m²)
pound-mass/foot ³		kilogram/meter ³
	1.601 846 X E +1	(kg/m ³)
rad (radiation dose absorbed)	1.000 000 X E -2	** Gray (Gy)
roentgen		coulomb/kilogram
	2.579 760 X E -4	(C/kg)
	1 000 000 1/ = 0	1

^{*} The becquerel (Bq) is the SI unit of radioactivity; 1 Bq = 1 event/s.

shake

torr (mm HG, O°C)

slug

A more complete listing of conversions may be found in "Metric Practice Guide E 380-74." American Society for Testing and Materials.

1.000 000 X E -8

1.459 390 X E +1

1.333 22 X E -1

second (s)

kilogram (kg)

kilo pascal (kPa)

^{**} The Gray (GY) is the SI unit of absorbed radiation.

INTENTIONALLY LEFT BLANK.

- 2 ADMINISTRATOR
 ATTN DTIC DDA
 DEFENSE TECHNICAL INFO CTR
 CAMERON STATION
 ALEXANDRIA VA 22304-6145
- 1 DIRECTOR
 ATTN AMSRL OP SD TA
 US ARMY RESEARCH LAB
 2800 POWDER MILL RD
 ADELPHI MD 20783-1145
- 3 DIRECTOR
 ATTN AMSRL OP SD TL
 US ARMY RESEARCH LAB
 2800 POWDER MILL RD
 ADELPHI MD 20783-1145
- 1 DIRECTOR
 ATTN AMSRL OP SD TP
 US ARMY RESEARCH LAB
 2800 POWDER MILL RD
 ADELPHI MD 20783-1145

ABERDEEN PROVING GROUND

5 DIR USARL ATTN AMSRL OP AP L (305)

- 2 HQDA
 ATTN SARD TR MS K KOMINOS
 DR R CHAIT
 PENTAGON
 WASHINGTON DC 20310-0103
- 2 HQDA
 ATTN SARD TT MS C NASH
 DR F MILTON
 PENTAGON
 WASHINGTON DC 20310-0103
- 1 DIR OF DEFNS RSRCH AND ENGRG ATTN DD TWP WASHINGTON DC 20301
- 1 ASST SECRETARY OF DEFNS
 ATTN DOCUMENT CONTROL
 ATOMIC ENERGY
 WASHINGTON DC 20301
- 1 CHAIRMAN
 ATTN J 5 R&D DIV
 JOINT CHIEFS OF STAFF
 WASHINGTON DC 20301
- 2 DA DCSOPS ATTN TECH LIB DIR OF CHEM & NUC OPS WASHINGTON DC 20310
- 1 EUROPEAN RSRCH OFC ATTN DR R REICHENBACH USARDSG UK PSC 802 BOX 15 FPO AE 09499-1500
- 1 DIR
 ATTN TECH LIB
 ADVNCD RSRCH PROJ AGCY
 3701 N FAIRFAX DR
 ARLINGTON VA 22203-1714
- 2 CDR
 ATTN AMSEL RD
 AMSEL RO TPPO P
 US ARMY CECOM
 FT MONMOUTH NJ 07703-5301
- 1 MIT ATTN TECH LIB CAMBRIDGE MA 02139

- 2 DIR
 ATTN PUBLIC RELATIONS OFC
 TECH LIB
 FED EMERG MGMT AGCY
 WASHINGTON DC 20472
- 1 CHAIRMAN
 DOD EXPLOSIVES SAFETY BOARD
 ROOM 856 C HOFFMAN BLDG 1
 2461 EISENHOWER AVE
 ALEXANDRIA VA 22331-0600
- 1 DIR
 ATTN DT 2 WPNS & SYS DIV
 DEFNS INTLLGNC AGCY
 WASHINGTON DC 20301
- 8 DIR
 ATTN CSTI TECH LIB
 DDIR
 DFSP
 NANS
 OPNA
 SPSD
 SPTD
 DFTD
 DEFNS NUCLEAR AGENCY
 WASHINGTON DC 20305
- 3 CDR
 ATTN FCPR
 FCTMOF
 NMHE
 FIELD COMMAND DNA
 KIRTLAND ARB NM 87115
- 10 CIA
 ATTN GE 47 HQ
 DIR DB STANDARD
 WASHINGTON DC 20505
- 2 CDR
 ATTN AMSNA D DR D SIELING
 STRNC UE J CALLIGEROS
 US ARMY NRDEC
 NATICK MA 01762
- 1 CDR
 ATTN ASQNC ELC IS L R MYER CTR
 US ARMY CECOM
 R&D TECH LIB
 FT MONMOUTH NJ 07703-5000

- 1 CDR
 ATTN SMCAR FSM W BARBER BLDG 94
 US ARMY ARDEC
 PCTNY ARSNL NJ 07806-5000
- 1 DIR
 ATTN AIAMS YDL
 US ARMY MISSILE & SPACE INTLLGNC CTR
 REDSTONE ARSNL AL 35898-5500
- 1 DIR ATTN AMSMR ATL US ARMY RESEARCH LAB WATERTOWN MA 02172-0001
- 1 CDR
 ATTN HNDED FD
 US ARMY ENGINEER DIV
 PO BOX 1500
 HUNTSVILLE AL 35807
- 1 CDR
 ATTN CESWF PM J
 US ARMY CORPS OF ENGRS
 FT WORTH DISTRICT
 PO BOX 17300
 FT WORTH TX 76102-0300
- 1 CDR
 ATTN SLCRO D
 US ARMY RESEARCH OFFICE
 PO BOX 12211
 RSCH TRI PK NC 27709-2211
- 1 DIR
 ATTN ATRC RPR RADDA
 HQ TRAC RPD
 FT MONROE VA 23651-5143
- 1 DIR ATTN ATRC WC KIRBY TRAC WSMR WSMR NM 88002-5502
- 1 CDR
 ATTN STEWS NED DR MEASON
 US ARMY WSMR
 WSMR NM 88002-5158

- 2 CHIEF OF NAVAL OPERATIONS
 ATTN OP 03EG
 OP 985F
 DEPT OF THE NAVY
 WASHINGTON DC 20350
- 1 CDR
 ATTN RSRCH AND DATA BRANCH
 US ARMY NGIC
 220 7TH STREET NE
 CHARLOTTESVILLE VA 22901-5396
- 1 DIR
 ATTN ATRC L MR CAMERON
 US ARMY TRAC FT LEE
 FORT LEE VA 23801-6140
- 2 CDR
 ATTN CSSD H MPL TECH LIB
 CSSD H XM DR DAVIES
 US ARMY STRATEGIC DEFENSE COMMAND
 PO BOX 1500
 HUNTSVILLE AL 35807
- 3 CDR
 ATTN CEWES SS R J WATT
 CEWES SE R J INGRAM
 CEWES TL TECH LIBRARY
 US ARMY CORPS OF ENGINEERS
 WATERWAYS EXPERIMENT STATION
 PO BOX 631
 VICKSBURG MS 39180-0631
- 3 CDR
 US ARMY NUCLEAR & CHEMICAL AGENCY
 7150 HELLER LOOP, SUITE 101
 SPRINGFIELD VA 22150-3198
- 1 DIR
 ATTN ATRC
 TRAC FLVN
 FORT LEAVENWORTH KS 66027-5200
- 1 CDR
 ATTN PME 117 21A
 NAVAL ELECTRONIC SYSTEMS COMMAND
 WASHINGTON DC 20360
- 2 OFFICE OF NAVAL RESEARCH ATTN DR A FAULSTICK CODE 23 800 N QUINCY STREET ARLINGTON VA 22217

- 1 CDR
 ATTN CODE SEA 62R
 NAVAL SEA SYSTEMS COMMAND
 DEPARTMENT OF THE NAVY
 WASHINGTON DC 20362-5101
- 1 COMMANDING OFFICER CODE L51
 ATTN J TANCRETO
 NAVAL CIVIL ENGINEERING LABORATORY
 PORT HUENEME CA 93043-5003
- 1 CIVIL ENGINEERING LABORATORY
 ATTN TECHNICAL LIBRARY CODE L31
 NAVAL CONSTRUCTION BATTALION CTR
 PORT HUENEME CA 93041
- 1 CDR
 ATTN CODE E23 LIBRARY
 NAVAL SURFACE WARFARE CENTER
 DAHLGREN VA 22448-5000
- 1 WHITE OAK WARFARE CTR DETACHMENT ATTN CODE E232 TECHNICAL LIBRARY 10901 NEW HAMPSHIRE AVENUE SILVER SPRING MD 20903-5000
- 1 CDR
 ATTN DOCUMENT CONTROL
 NAVAL WEAPONS EVALUATION FAC
 KIRTLAND AFB NM 87117
- 1 AEDC ATTN R MCAMIS MAIL STOP 980 ARNOLD AFB TN 37389
- 1 OLAC PL TSTL ATTN D SHIPLETT EDWARDS AFB CA 93523-5000
- 2 AIR FORCE ARMAMENT LABORATORY
 ATTN AFATL DOIL
 AFATL DLYV
 EGLIN AFB FL 32541-5000
- 3 PHILLIPS LABORATORY (AFWL)
 ATTN NTE
 NTED
 NTES
 KIRTLAND AFB NM 87117-6008

- 1 AFIT
 ATTN TECHNICAL LIBRARY
 BLDG 640 B
 WRIGHT PATTERSON AFB OH 45433
- 1 FTD NIIS WRIGHT PATTERSON AFB OH 45433
- 4 DIR
 ATTN R GUENZLER MS 3505
 R HOLMAN MS-3510
 R A BERRY
 W C REED
 IDAHO NATIONAL ENGINEERING LABORATORY
 EG&G IDAHO INC
 PO BOX 8757 BWI AIRPORT
 BALTIMORE MD 21240
- 3 KAMAN SCIENCES CORPORATION
 ATTN LIBRARY
 P A ELLIS
 F H SHELTON
 P O BOX 7463
 COLORADO SPRINGS CO 80933-7463
- 2 DIR
 ATTN TH DOWLER MS F602
 DOC CONTROL FOR REPORTS LIBRARY
 LOS ALAMOS NATIONAL LABORATORY
 PO BOX 1663
 LOS ALAMOS NM 87545
- 1 DIR
 ATTN DOC CONTROL FOR TECH LIB
 SANDIA NATIONAL LABORATORIES
 LIVERMORE LABORATORY
 P O BOX 969
 LIVERMORE CA 94550
- 1 DIR
 ATTN TECHNICAL LIBRARY
 NASA LANGLEY RESEARCH CENTER
 HAMPTON VA 23665
- 1 ADA TECHNOLOGIES INC ATTN JAMES R BUTZ HONEYWELL CENTER SUITE 110 304 INVERNESS WAY SOUTH ENGLEWOOD CO 80112

- 1 ALLIANT TECHSYSTEMS INC ATTN ROGER A RAUSCH MN48 3700 7225 NORTHLAND DRIVE BROOKLYN PARK MN 55428
- 1 AEROSPACE CORPORATION ATTN TECH INFO SERVICES P O BOX 92957 LOS ANGELES CA 90009
- 1 THE BOEING COMPANY
 ATTN AEROSPACE LIBRARY
 P O BOX 3707
 SEATTLE WA 98124
- 1 CALIFORNIA RES & TECH INC ATTN M ROSENBLATT 20943 DEVONSHIRE STREET CHATSWORTH CA 91311
- 1 DYNAMICS TECHNOLOGY INC ATTN D T HOVE G P MASON 21311 HAWTHORNE BLVD SUITE 300 TORRANCE CA 90503
- 1 EATON CORPORATION
 ATTN J WADA
 DEFENSE VALVE & ACTUATOR DIV
 2338 ALASKA AVE
 EL SEGUNDO CA 90245-4896
- 5 DIR
 ATTN DOC CONTROL 3141
 C CAMERON DIV 6215
 A CHABAI DIV 7112
 D GARDNER DIV 1421
 J MCGLAUN DIV 1541
 SANDIA NATIONAL LABORATORIES
 P O BOX 5800
 ALBUQUERQUE NM 87185-5800
- 1 BLACK & VEATCH ENGINEERS - ARCHITECTS ATTN H D LAVERENTZ 1500 MEADOW LAKE PARKWAY KANSAS CITY MO 64114

- 1 DIRECTOR
 ATTN DR T HOLTZ MS 202-14
 NASA-AMES RESEARCH CENTER
 APPLIED COMPUTATIONAL AERO BRANCH
 MOFFETT FIELD CA 94035
- 2 APPLIED RESEARCH ASSOCIATES INC ATTN J KEEFER N H ETHRIDGE P O BOX 548 ABERDEEN MD 21001
- 1 CARPENTER RESEARCH CORPORATION ATTN H JERRY CARPENTER 27520 HAWTHORNE BLVD SUITE 263 ROLLING HILLS ESTATES CA 90274
- 1 GOODYEAR CORPORATION ATTN R M BROWN BLDG 1 SHELTER ENGINEERING LITCHFIELD PARK AZ 85340
- 2 FMC CORPORATION
 ATTN J DROTLEFF
 C KREBS MDP95
 ADVANCED SYSTEMS CENTER
 BOX 58123
 2890 DE LA CRUZ BLVD
 SANTA CLARA CA 95052
- 1 SVERDRUP TECHNOLOGY INC ATTN B D HEIKKINEN SVERDRUP CORPORATION AEDC MS 900 ARNOLD AFB TN 37389-9998
- 1 KTECH CORPORATION
 ATTN DR E GAFFNEY
 901 PENNSYLVANIA AVE NE
 ALBUQUERQUE NM 87111
- 4 KAMAN AVIDYNE
 ATTN R RUETENIK (2 CYS)
 S CRISCIONE
 R MILLIGAN
 83 SECOND AVENUE
 NORTHWEST INDUSTRIAL PARK
 BURLINGTON MA 01830

- 2 KAMAN SCIENCES CORPORATION ATTN DASIAC (2 CYS) P O DRAWER 1479 816 STATE STREET SANTA BARBARA CA 93102-1479
- 1 LOCKHEED MISSILES & SPACE CO ATTN J J MURPHY DEPT 81 11 BLDG 154 P O BOX 504 SUNNYVALE CA 94086
- 1 ORLANDO TECHNOLOGY INC ATTN D MATUSKA 60 SECOND STREET BLDG 5 SHALIMAR FL 32579
- 2 THE RALPH M PARSONS COMPANY ATTN T M JACKSON LB TS PROJECT MANAGER 100 WEST WALNUT STREET PASADENA CA 91124
- 1 SAIC
 ATTN N SINHA
 501 OFFICE CENTER DRIVE APT 420
 FT WASHINGTON PA 19034 3211
- 1 SAIC
 ATTN J GUEST
 2301 YALE BLVD SE
 SUITE E
 ALBUQUERQUE NM 87106
- 2 S CUBED
 A DIVISION OF MAXWELL LABS INC
 ATTN C E NEEDHAM
 L KENNEDY
 2501 YALE BLVD SE
 ALBUQUERQUE NM 87106
- 1 TRW
 BALLISTIC MISSILE DIVISION
 ATTN H KORMAN
 MAIL STATION 526 614
 P O BOX 1310
 SAN BERNADINO CA 92402
- 1 THERMAL SCIENCE INC ATTN R FELDMAN 2200 CASSENS DR ST LOUIS MO 63026

- 2 MCDONNELL DOUGLAS ASTRNTCS CORP ATTN ROBERT W HALPRIN K A HEINLY 5301 BOLSA AVENUE HUNTINGTON BEACH CA 92647
- 1 MDA ENGINEERING INC ATTN DR DALE ANDERSON 500 EAST BORDER STREET SUITE 401 ARLINGTON TX 07601
- 2 PHYSICS INTERNATIONAL CORPORATION P O BOX 5010 SAN LEANDRO CA 94577-0599
- 1 R&D ASSOCIATES ATTN G P GANONG P O BOX 9377 ALBUQUERQUE NM 87119
- 1 SCIENCE CENTER
 ROCKWELL INTERNATIONAL CORPORATION
 ATTN DR S CHAKRAVARTHY
 DR D OTA
 1049 CAMINO DOS RIOS
 THOUSAND OAKS CA 91358
- 3 S CUBED
 A DIVISION OF MAXWELL LABS INC
 ATTN TECHNICAL LIBRARY
 R DUFF
 K PYATT
 P O BOX 1620
 LA JOLLA CA 92037-1620
- 1 SUNBURST RECOVERY INC ATTN DR C YOUNG P O BOX 2129 STEAMBOAT SPRINGS CO 80477
- 1 SVERDRUP TECHNOLOGY INC ATTN R F STARR P O BOX 884 TULLAHOMA TN 37388
- 1 SRI INTERNATIONAL
 ATTN DR G R ABRAHAMSON
 DR J GRAN
 DR B HOLMES
 333 RAVENWOOD AVENUE
 MENLO PARK CA 94025

- 1 BATTELLE
 TWSTIAC
 505 KING AVENUE
 COLUMBUS OH 43201-2693
- 2 THINKING MACHINES CORPORATION ATTN G SABOT R FERREL 245 FIRST STREET CAMBRIDGE MA 02142-1264
- 1 CALIFORNIA INSTITUTE OF TECHNOLOGY ATTN T J AHRENS 1201 E CALIFORNIA BLVD PASADENA CA 91109
- 1 UNIVERSITY OF MINNESOTA
 ARMY OF HIGH PERF COMP RES CTR
 ATTN DR TAYFUN E TEZDUYAR
 1100 WASHINGTON AVE SOUTH
 MINNEAPOLIS MN 55415
- 2 CDR
 ATTN SSCNC YSD J ROACH
 SSCNC WST A MURPHY
 US ARMY NRDEC
 KANSAS STREET
 NATICK MA 10760-5018
- 3 SOUTHWEST RESEARCH INSTITUTE
 ATTN DR C ANDERSON
 S MULLIN
 A B WENZEL
 P O DRAWER 28255
 SAN ANTONIO TX 78228-0255
- 1 STATE UNIVERSITY OF NEW YORK MECHANICAL & AEROSPACE ENGRG ATTN DR PEYMAN GIVI BUFFALO NY 14260
- 2 DENVER RESEARCH INSTITUTE ATTN J WISOTSKI TECHNICAL LIBRARY P O BOX 10758 DENVER CO 80210
- 2 UNIVERSITY OF MARYLAND
 INSTITUTE FOR ADV COMPUTER STUDIES
 ATTN L DAVIS
 G SOBIESKI
 COLLEGE PARK MD 20742

NO. OF COPIES ORGANIZATION

- 1 NORTHROP UNIVERSITY
 ATTN DR F B SAFFORD
 5800 W ARBOR VITAE STREET
 LOS ANGELES CA 90045
- 1 STANFORD UNIVERSITY ATTN DR D BERSHADER DURAND LABORATORY STANFORD CA 94305

ABERDEEN PROVING GROUND

- 1 CDR USATACOM ATTN AMSTE-TE-F (L TELETSKI)
- 1 CDR USATHAMA ATTN AMSTH-TE
- 1 CDR USACSTA ATTN STEC-LI
- DIR USARL
 ATTN AMSRL-SL-CM (BLDG E3331)
 AMSRL-WT-NC (R LOTTERO)(2 CP)
 AMSRL-WT-NC (A. MIHALCIN)(2 CP)

INTENTIONALLY LEFT BLANK.

USER EVALUATION SHEET/CHANGE OF ADDRESS

This Laboratory undertakes a continuing effort to improve the quality of the reports it publishes. Your comments/answers to the items/questions below will aid us in our efforts. 1. ARL Report Number ARL-CR-235 Date of Report July 1995 2. Date Report Received _____ 3. Does this report satisfy a need? (Comment on purpose, related project, or other area of interest for which the report 4. Specifically, how is the report being used? (Information source, design data, procedure, source of ideas, etc.) 5. Has the information in this report led to any quantitative savings as far as man-hours or dollars saved, operating costs avoided, or efficiencies achieved, etc? If so, please elaborate. 6. General Comments. What do you think should be changed to improve future reports? (Indicate changes to organization, technical content, format, etc.) Organization **CURRENT** Name **ADDRESS** Street or P.O. Box No. City, State, Zip Code 7. If indicating a Change of Address or Address Correction, please provide the Current or Correct address above and the Old or Incorrect address below. Organization OLD Name **ADDRESS** Street or P.O. Box No.

(Remove this sheet, fold as indicated, tape closed, and mail.)
(DO NOT STAPLE)

City, State, Zip Code